
Basic 11

High Knudsen Number Flows

Prof. T. Niimi

Dept. of Micro/Nano Systems Engineering
Nagoya University



Basic 11 High Knudsen Number Flows
COE for Education and Research of Micro-Nano Mechatronics, Nagoya University

Prof. T. Niimi

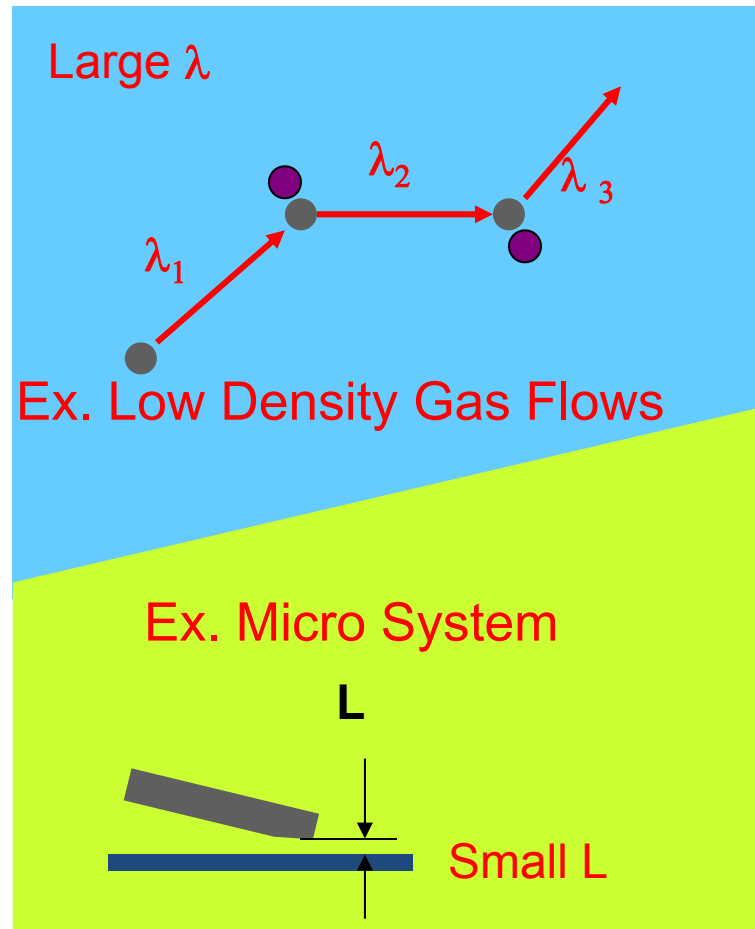


Contents

1. High Knudsen Number Flows
2. Development of Optical Diagnostics Methods for High Knudsen Number Flows
 - Laser Induced Fluorescence (LIF) :
 - Resonantly Enhanced Multi-Photon Ionization (REMPI)
 - Rotational Nonequilibrium in Low Density N₂ Jets
3. Application of Pressure Sensitive Paint (PSP)
 - to High Knudsen Number Flows
 - Development of PSP for Low Density Gas Flows
 - Development of PSMF for Micro- and Nano-Devices
 - (PSMF: Pressure Sensitive Molecular Film)
 - LB Method for Fabrication of PSMF
 - Application of PSMF to Micro-Flows
4. Applications of PSP in Atmospheric Condition
 - Rotating Disks
 - Mixing Chamber



High Knudsen Number Flows



Knudsen number

$$Kn = \lambda / L$$

λ : Mean Free Path

L: Characteristic Dimension

Continuum approach invalid

Kn larger than ~ 0.1

→ “High Kn Number Flow”
(Large λ or Small L)

Flow field is strongly influenced by interaction of molecules with a solid boundary rather than intermolecular collisions

Mean Free Path

Mean Free Path

Average distance that a molecule travels between successive collisions

$$\lambda = \frac{1}{\sqrt{2}\pi d^2 n}$$

d : diameter of a molecule
 n : number density

Loschmit's Number

The molecular number density of an ideal gas at standard condition [0°C, 1 atm(=760 Torr =760 mmHg)] in 1 cm³

$$n=2.68699 \times 10^{19} \text{ cm}^{-3}$$

Mean free path in atmospheric pressure condition

Air: single molecular gas, mean diameter of a molecule $d=3.7 \times 10^{-10} \text{ m}$

$$\rightarrow \lambda=6.1 \times 10^{-8} \text{ m (61 nm)}$$

From the equation of state $p=nkT \rightarrow$

$$\lambda \propto \frac{T}{p}$$

(p : pressure, T : Temp., k : Boltzmann's const.)

$\rightarrow \lambda$ is inversely proportional to P and proportional to T



Traditional High Knudsen Number Flow

Rarefied Gas Flow or Atomic/Molecular Gas Flow

Rarefied Gas Flow

Atomic/Molecular Gas Flow

- ◆ Continuum Approach invalid
 - Analyses by Boltzmann Eq.
 - Direct Simulation Monte Carlo (DSMC)
- ◆ Nonequilibrium Phenomena
- ◆ Strong Influence of Interface
 - Accommodation Coefficient
 - Adsorption Probability
 - Physical Models of Solid Surface

International Symposium on Rarefied Gas Dynamics (since 1958)

Rarefied Gas Dynamics (RGD) has developed along with space technologies from 1950's and expanded into analyses of low density gas flows in vacuum technologies (such as semiconductor film growth)



From Rarefied Gas Flow to High Knudsen Number Flow

In so-called nano-technologies, attention has been focused mainly on fabrication of devices, but not on gas-surface interaction which is important for the devices working in ambient gas.

Flows around the micro-devices are also in the category of “Rarefied gas flows”, but it is difficult to accept that because the devices work mainly in the atmospheric condition



“High Knudsen Number Flow”



High Knudsen Number Flows

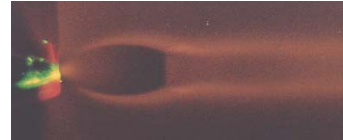


Technologies related to
Ultra-High Vacuum

Mean Free Path λ : large

High Kn Number Flows

Nano- and Micro-Devices
**Characteristic Dimension
 L : small**

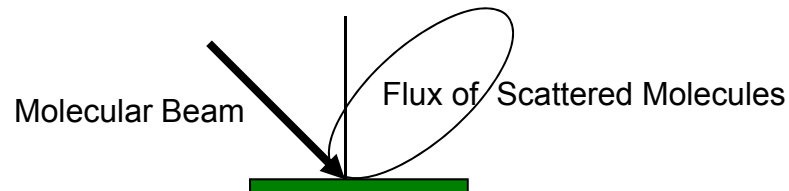


Strongly Nonequilibrium Gas Flows

Small Number of
Intermolecular Collisions
Flow field is strongly influenced
by interaction of molecules
with a solid boundary

Lack of Precise
Experimental Data

Gas-Surface Interaction



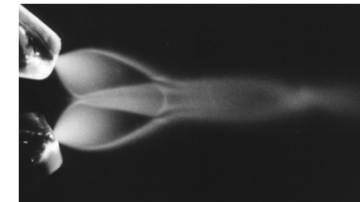
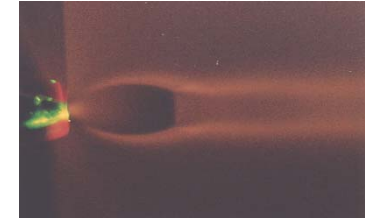
High Knudsen Number Flows Challenge of Our Group

Analyses of Flow Field Structures and Nonequilibrium Phenomena of the Low Density Gas Flows

Developments of Measurement Techniques using interaction of Laser Beams with Molecules

Measurement Techniques in Molecular Level:

LIF, CARS, DFWM, **REMPI**



Database Creation for Gas-Surface Interaction

Momentum and Energy Accommodation Coefficients

Experiments using a **Molecular Beam** and Detection of Reflected Gas Molecules including Internal Energy by use of **REMPI**

Pressure Distribution on Solid Surfaces in High Knudsen Number Flows

We can not Apply Pressure Taps to the Low Density Gas Flows and Micro- and Nano-Systems

→ Development of Measurement Techniques in Molecular Level
: **PSMF (Pressure Sensitive Molecular Film)**



Development of Optical Diagnostic Methods

for High Knudsen Number Flows

LIF

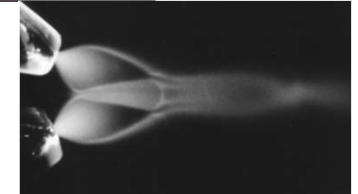
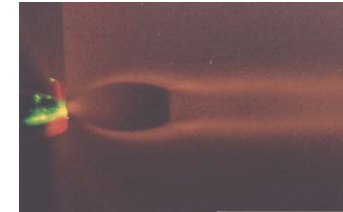
Laser Induced Fluorescence : I_2 , O_2 and NO

Flow visualization

Fujimoto, Niimi, Rarefied Gas Dynamics, AIAA(1988), 391-406

Niimi; Protokoll des DLR-Kolloquiums LIF-8, (1996), 62-68

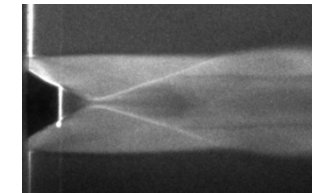
Mori, Niimi, et al; AIAA paper 2005-1350, AIAA, 2005-01(USA, Reno)



Temperature measurement technique

Niimi, et al; OPTICS LETTERS, 15-16(1990) , 918-920

Niimi, et al; Applied Optics, 34, 27(1995), 6275-6281



CARS

Coherent Anti-Stokes Raman Scattering : N_2

DFWM

Degenerate Four Wave Mixing : I_2

REMPI

Resonantly Enhanced Multi-Photon Ionization : N_2

Mori, Niimi, et al; Physics of Fluids, 17, 117103(2005)



Development of Optical Diagnostic Methods

for High Knudsen Number Flows

LIF

Laser Induced Fluorescence of I_2

Flow visualization of interacting supersonic free jets

Rotational temperature measurement technique

REMPI

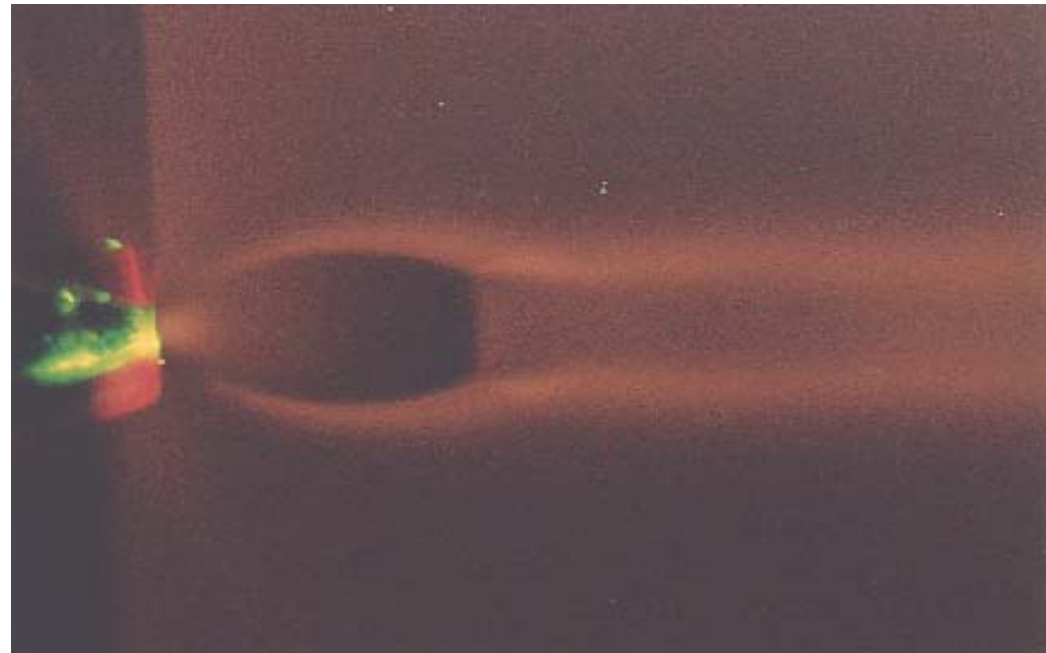
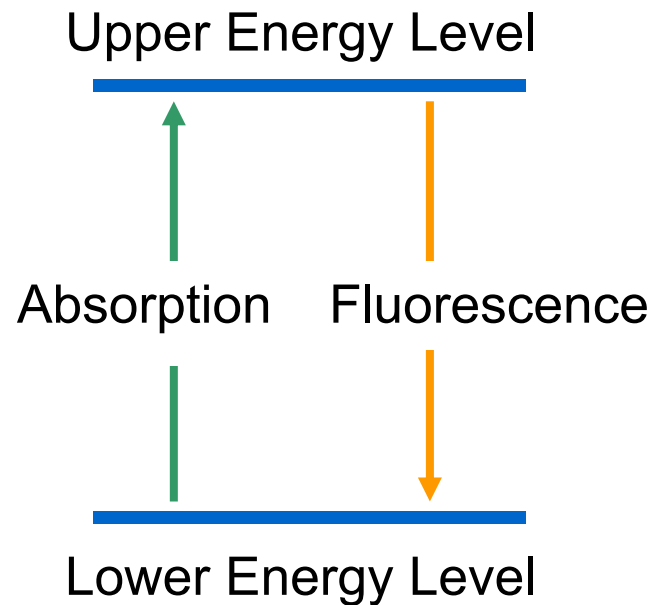
Resonantly Enhanced Multi-Photon Ionization : N_2

Application of REMPI to highly rarefied gas flows

Rotational nonequilibrium (non-Boltzmann distribution)



Laser Induced Fluorescence



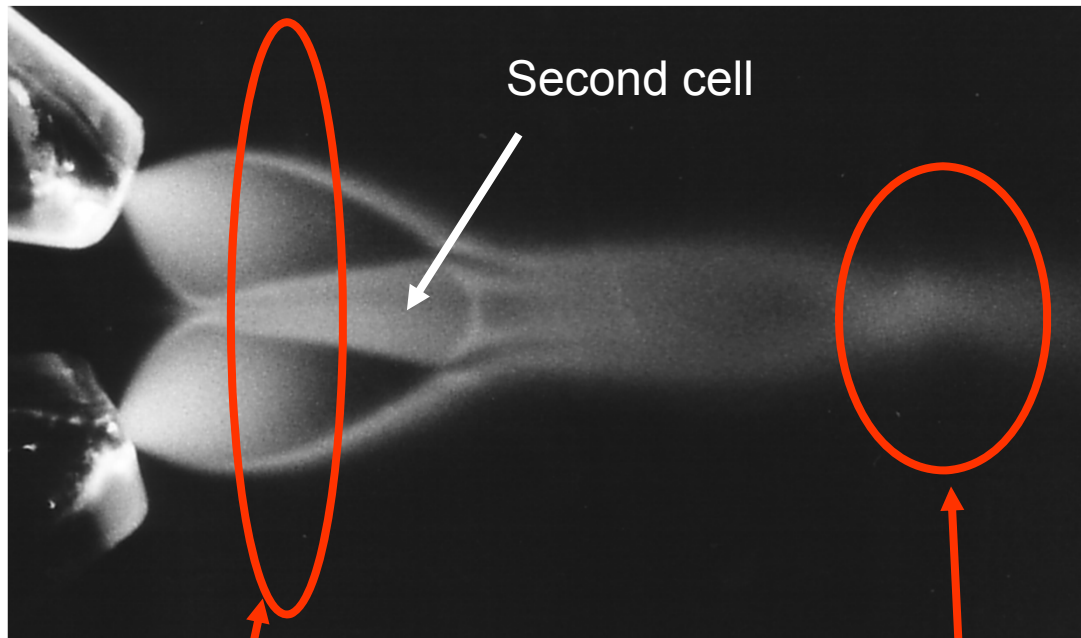
Principle of LIF

A supersonic Ar-free jet visualized by I₂-LIF
Nozzle Diameter: 0.5 mm
P_s/P_b= 150



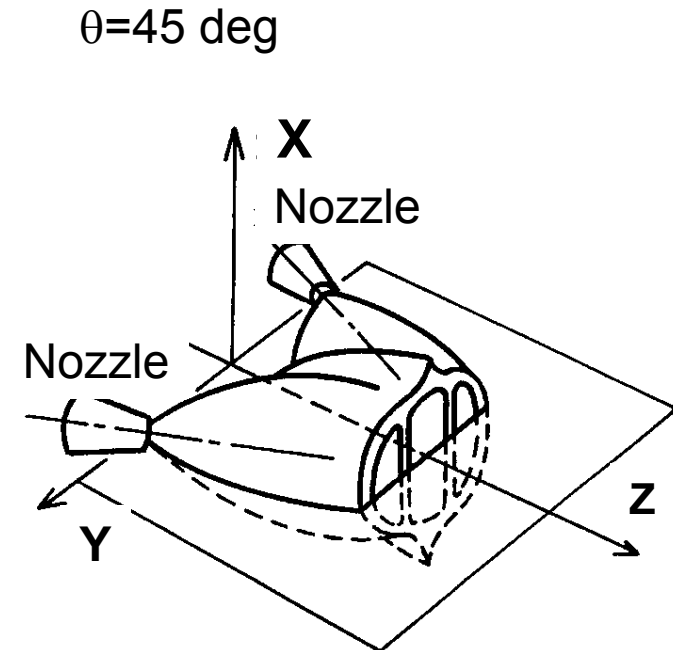
Application of I_2 -LIF to Flow Visualization

Flow field structure of two interacting supersonic free jets



ϕ -shaped structure

planar flow



$\theta = 45$ deg

X
Nozzle

Nozzle

Y

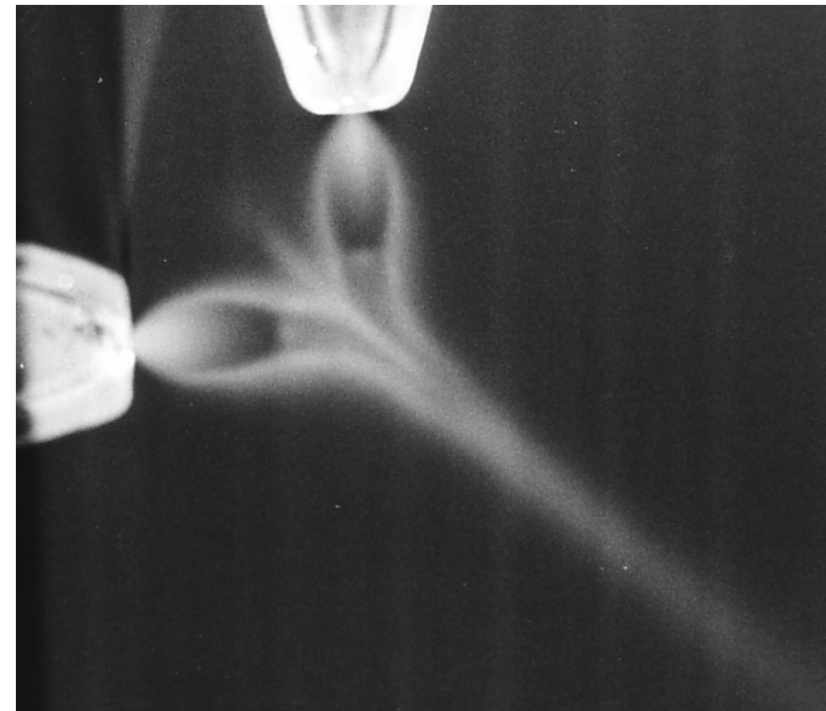
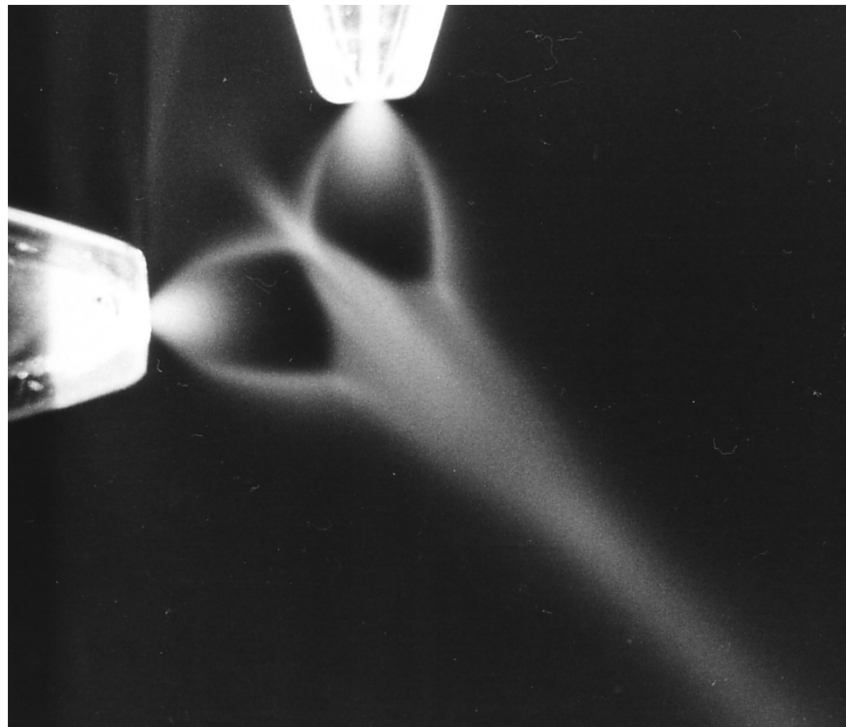
Z



Application of I₂-LIF to Flow Visualization

Flow field structure of two interacting supersonic free jets

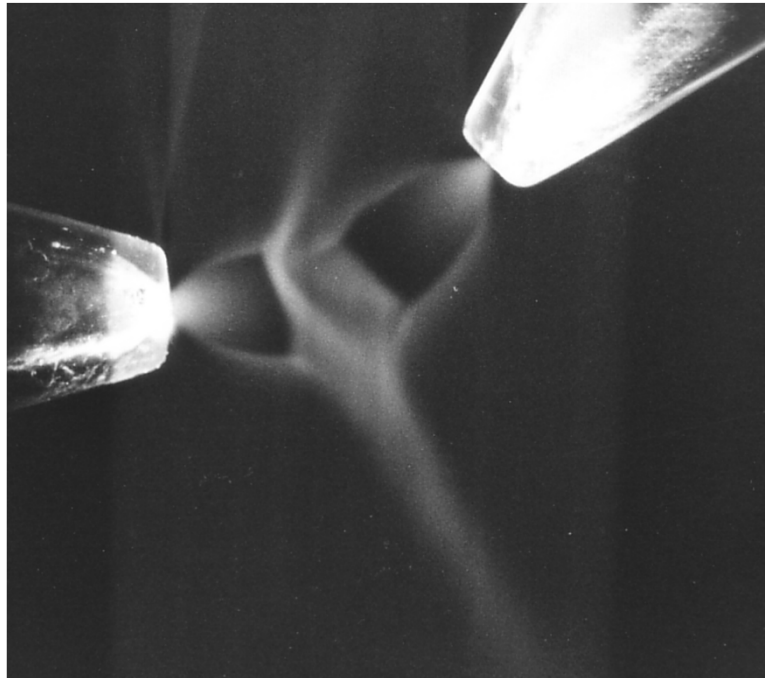
$\theta=90$ deg



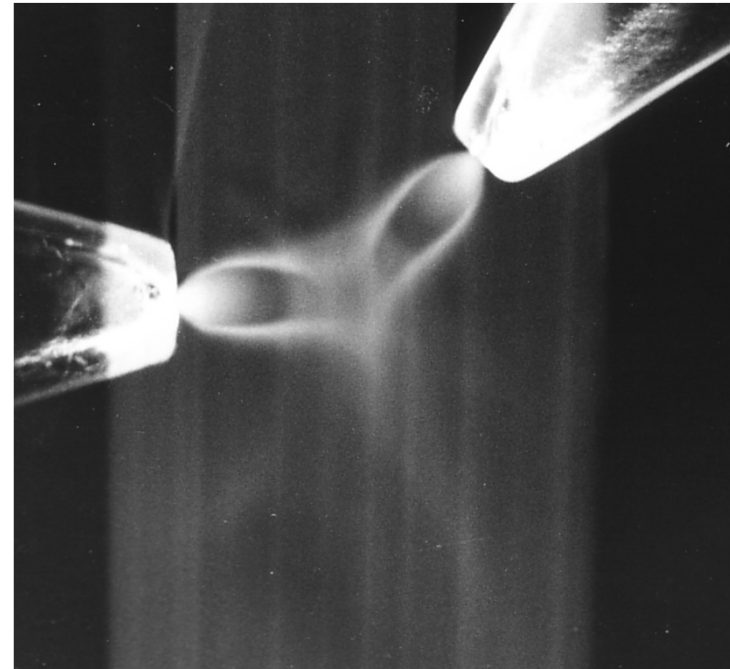
Application of I₂-LIF to Flow Visualization

Flow field structure of two interacting supersonic free jets

$\theta=135$ deg



stable



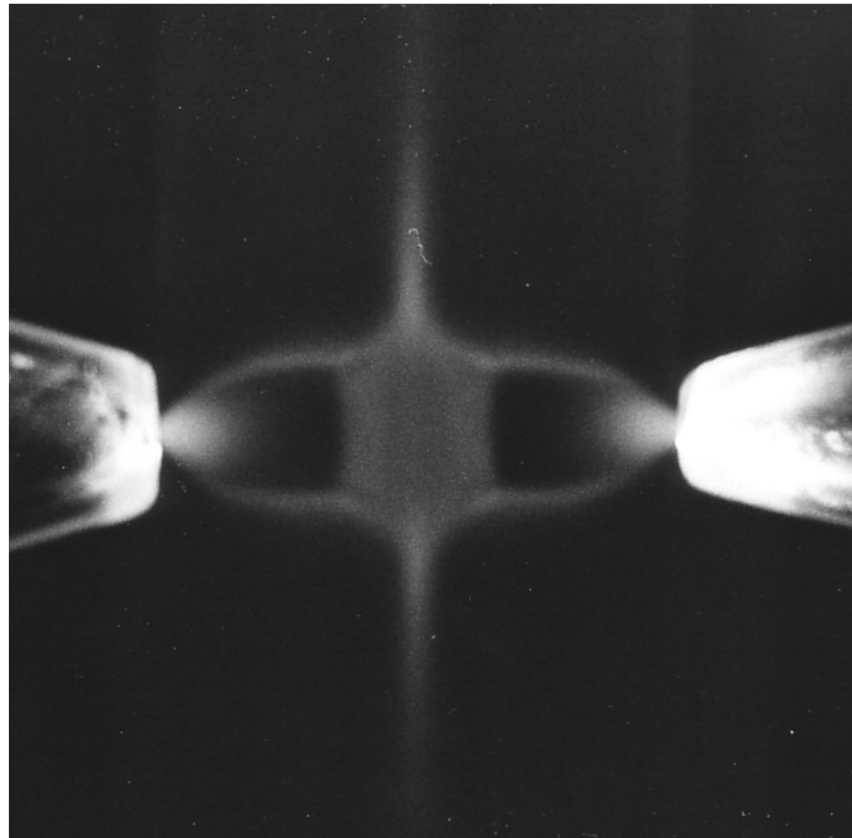
unstable



Application of I_2 -LIF to Flow Visualization

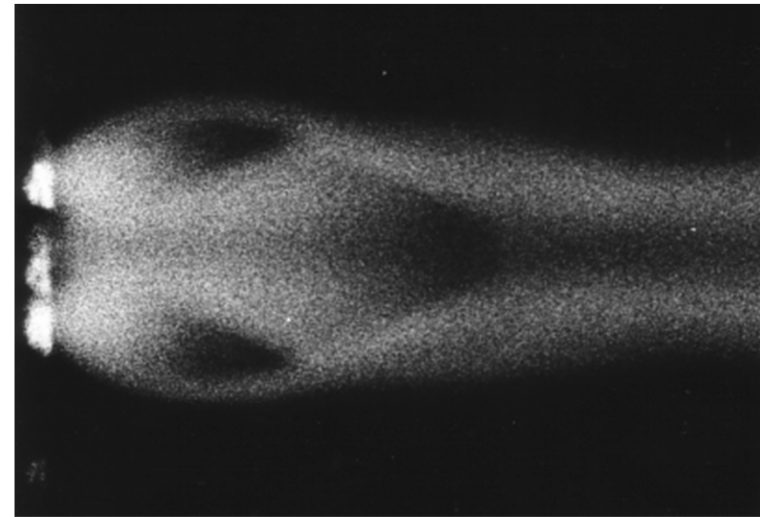
Flow field structure of two interacting supersonic free jets

$\theta=180$ deg

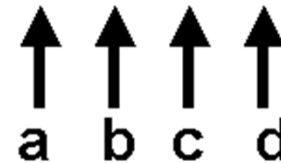


Application of I₂-LIF to Flow Visualization

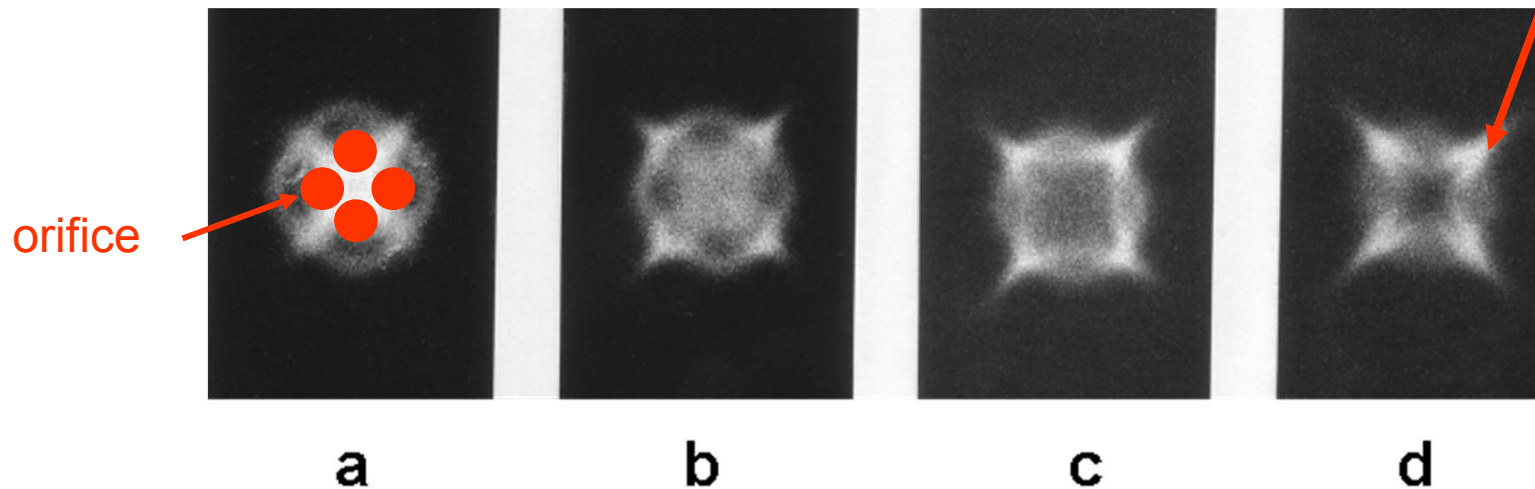
Flow field structure of
four interacting parallel
supersonic free jets



square arrangement of orifices



intense expansion



Rotational Temperature Measurement using LIF

Fluorescence intensity F (Two-Level Model)

$$F = C[A_{ji}/(A_{ji}+Q)]B_{ij}IfN_{I_2}$$

C : a constant, A_{ji} : spontaneous emission rate,

B_{ij} : stimulated-emission rate, Q : collision quenching rate,

I : intensity of laser beam, N_{I_2} : number density of I_2 ,

f : fraction of the ground-state population

Fluorescence intensity F_1 when the I_2 molecules in the rot. level J''_1 are excited

F_2 when the I_2 molecules in the rot. level J''_2 are excited

a ratio between these two fluorescence intensities

$$F_1/F_2 = [(B_{ij})_1 f_1] / [(B_{ij})_2 f_2],$$

If using common electronic and vibrational state for F_1 and F_2 ,

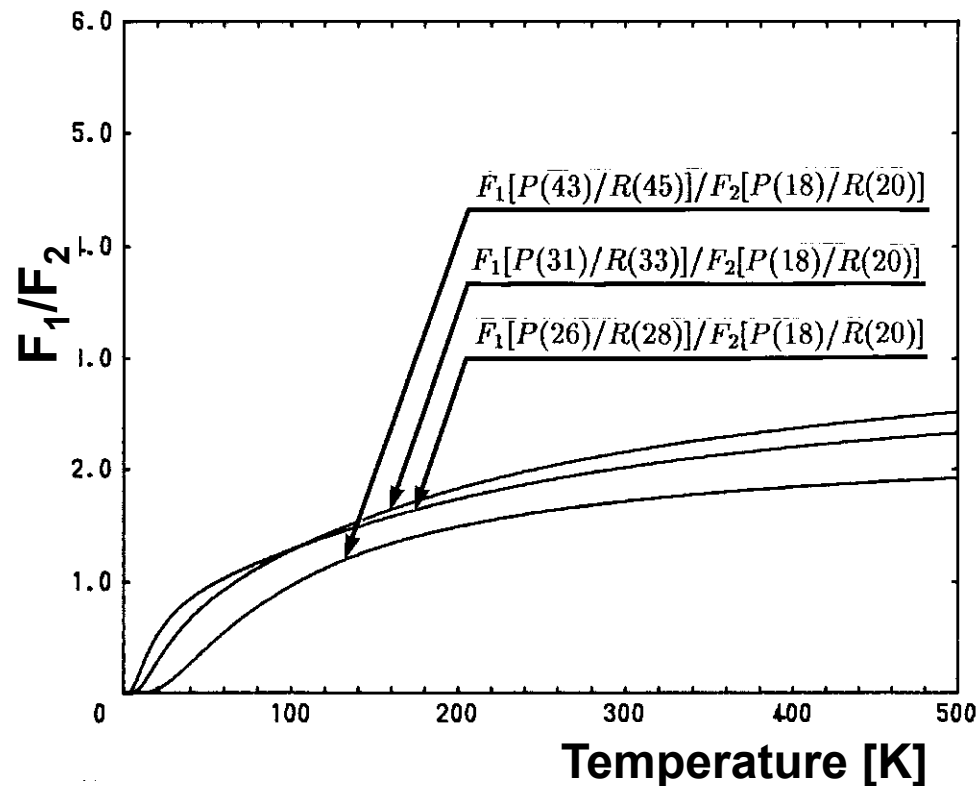
$$F_1/F_2 = [S(J''_1) f_r(J''_1, T)] / [S(J''_2) f_r(J''_2, T)]$$

S : Honl-London factor, f_r : fraction of the rotational population

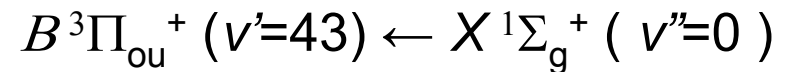
Once two lines are selected, the ratio can be expressed as a function of temp.



F₁/F₂ - Temperature



Absorption lines of I₂ molecules
in the transition of

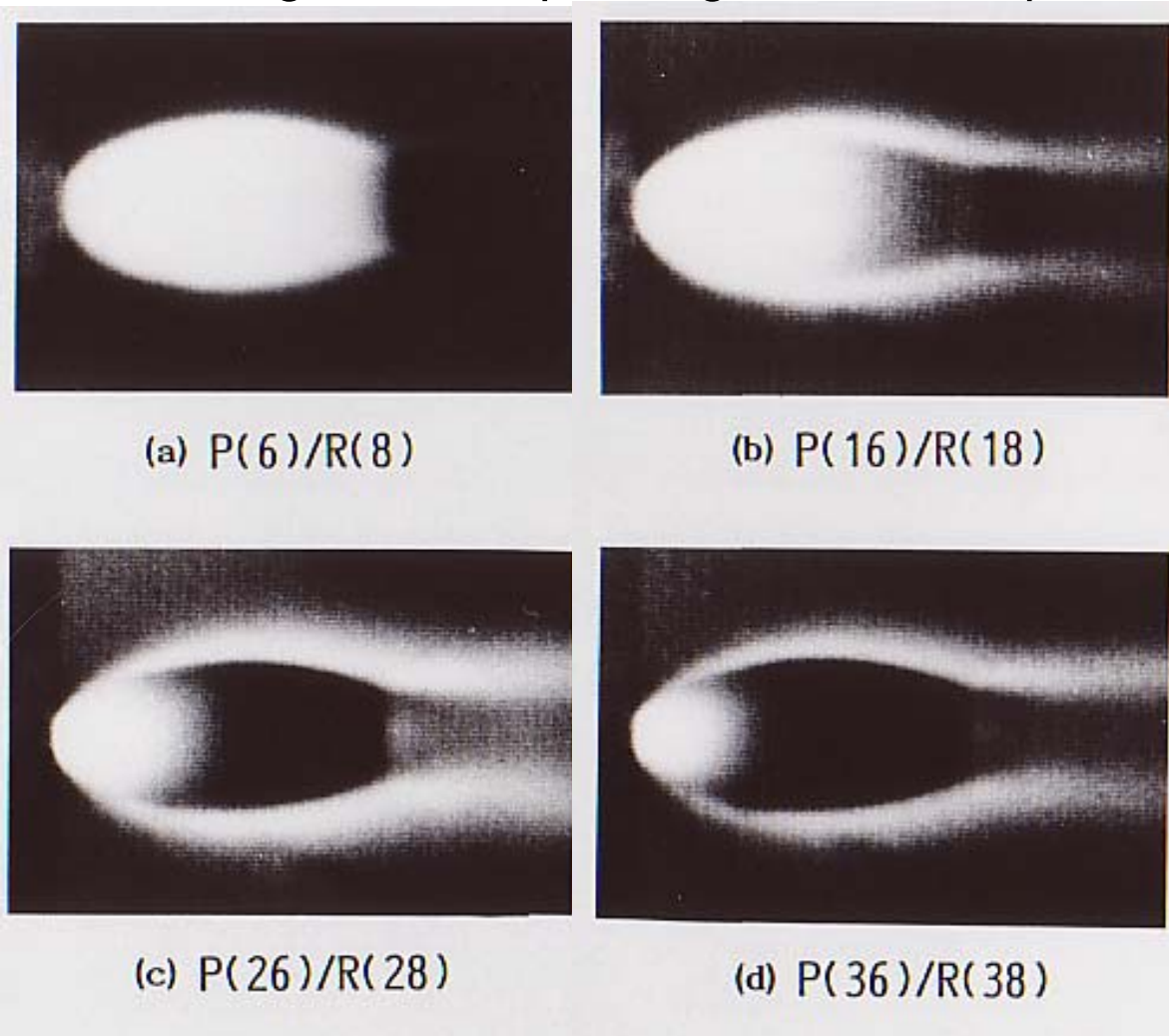


Wave Number	Absorption Line
19432.0415	P(8)/R(10)
28.0283	P(16)/R(18)
26.6531	P(18)/R(20)
19.6717	P(26)/R(28)
14.0906	P(31)/R(33)
396.8701	P(43)/R(45)



Fluorescence Intensity distribution depending on absorption lines

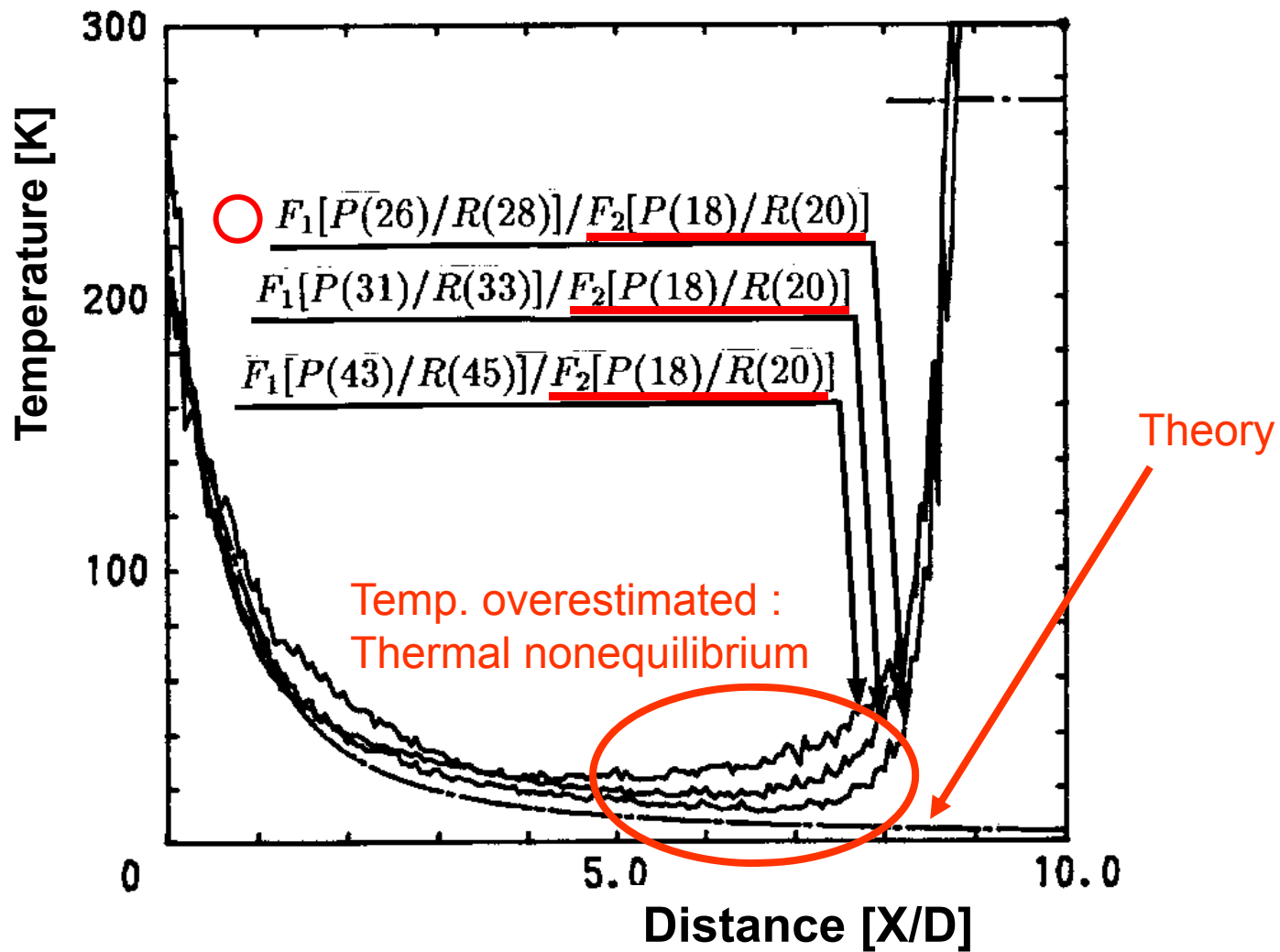
Supersonic free jets visualized by the use of irradiation of laser beams at wavelengths corresponding to the absorption lines



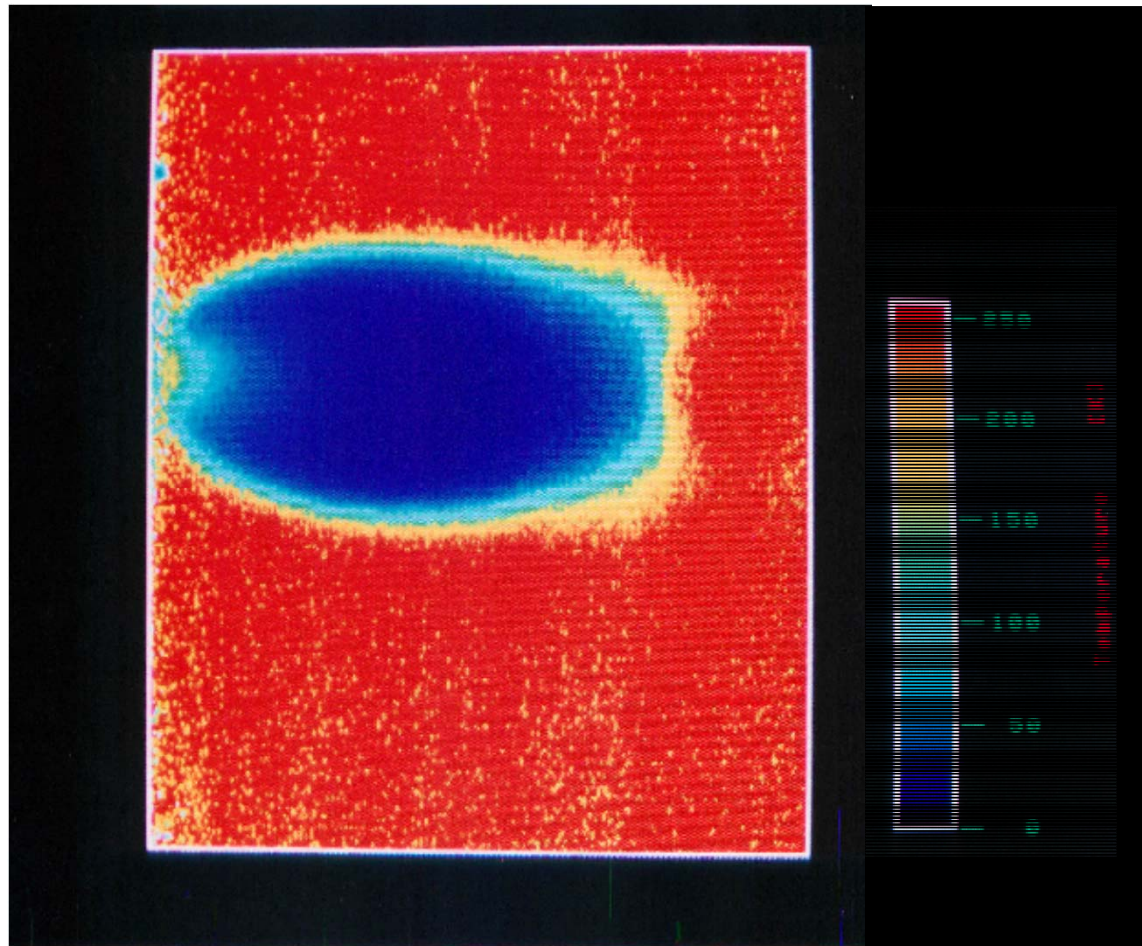
$P_s = 16 \text{ kPa}$
 $P_b = 100 \text{ Pa}$



Rotational Temperature distribution along the centerline of a Supersonic Free Jet



Rotational Temperature Distribution of a Supersonic Free Jet



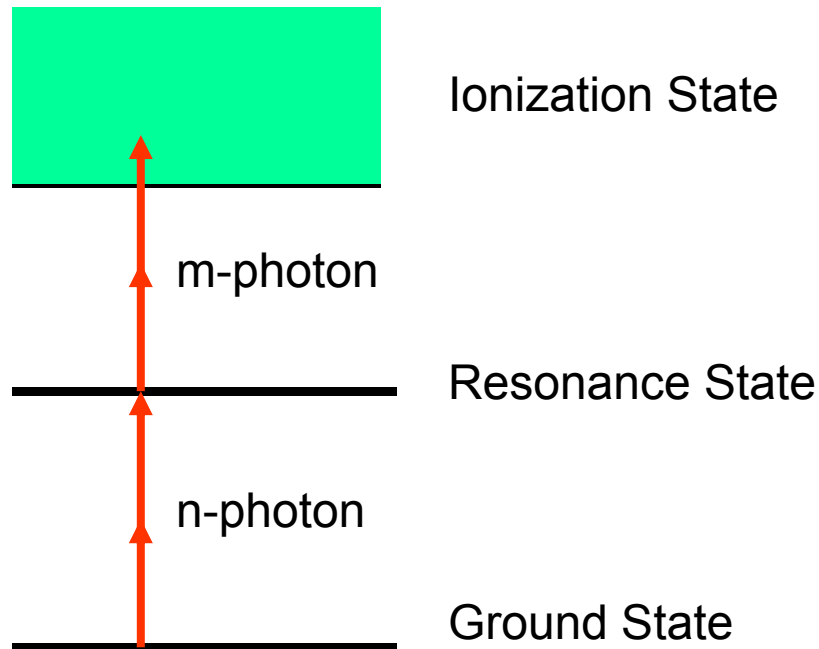
T. Niimi, et al, *Optics Letters*, 15-16(1990), 918-920
T. Niimi, et al, *Applied Optics*, 34, 27(1995), 6275-6281



REMPI

Resonantly Enhanced Multi-Photon Ionization

Non-intrusive Measurement Technique with High Sensitivity and Short Response, allowing measurement of non-equilibrium phenomena in the highly rarefied gas flows



$nR+m$ REMPI Process

Multi-Photon Ionization (MPI)
Low Ionization Rate
(by 1 step)



REMPI
High Ionization Rate
(by 2 steps)

- Short Response
 - High Sensitivity
- for $2R+2N_2$ -REMPI
 10^9 molecules/cm³



Objective of Our Study

Experimental detection of the non-Boltzmann distribution of the rotational levels (strongly nonequilibrium), applying the REMPI (Resonantly Enhanced Multi-Photon Ionization) method to the supersonic nitrogen free jets with P_0D of 15 Torr-mm or lower.

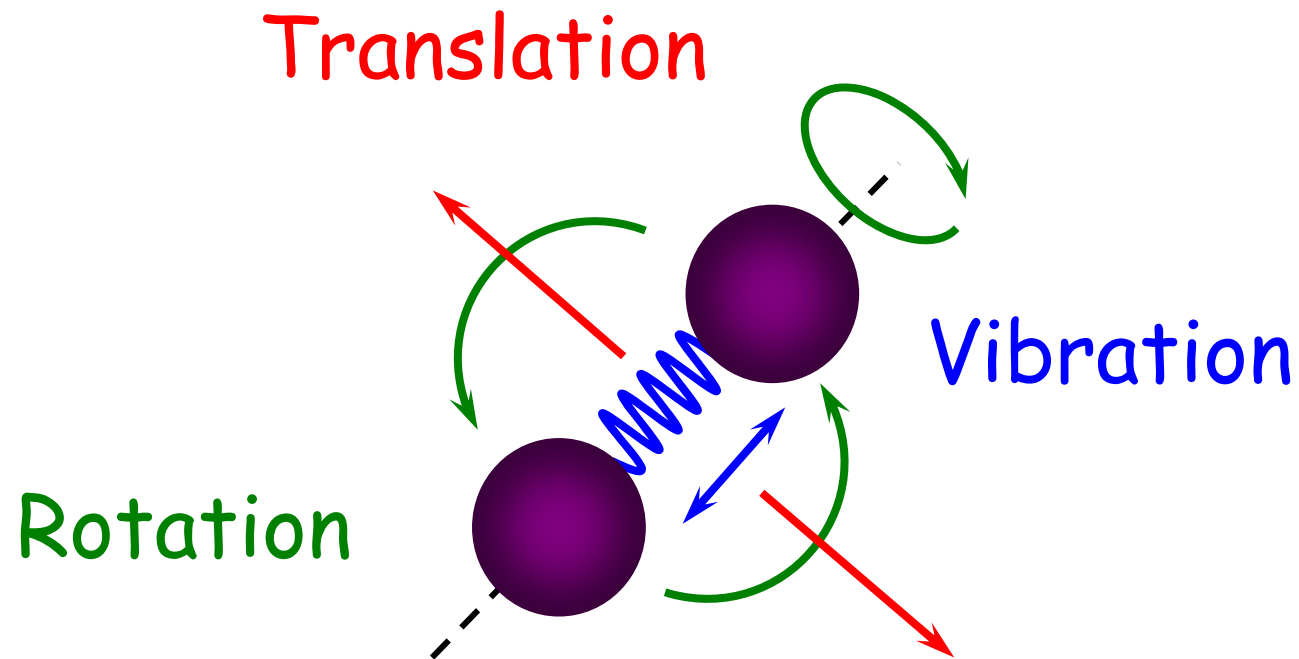
P_0D : Parameter depending inversely on nozzle Knudsen number

P_0 : stagnation pressure

D : orifice diameter



Modes of Motion for Diatomic Molecule



Distributions for translational and rotational energies

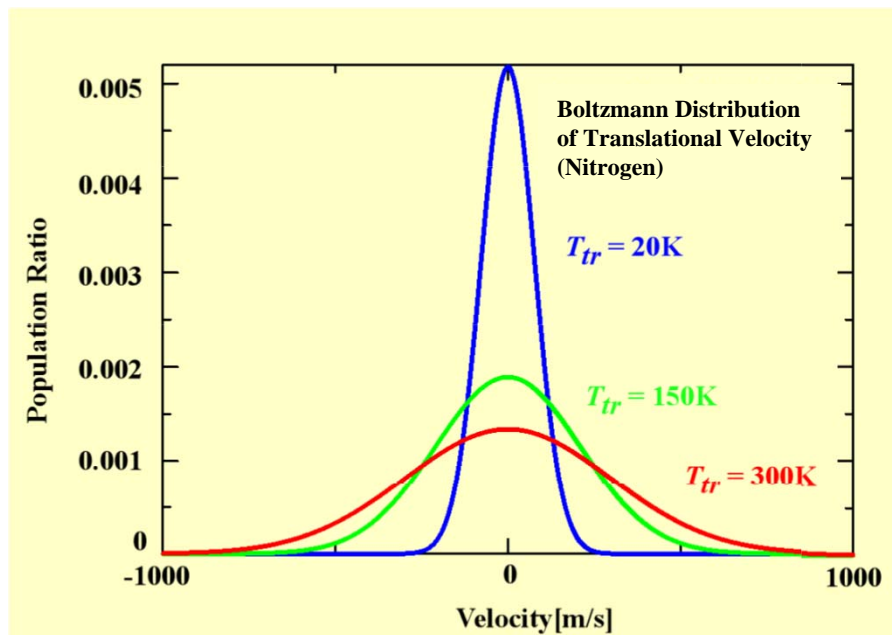
Boltzmann distributions (thermodynamic equilibrium)

Translational Velocity

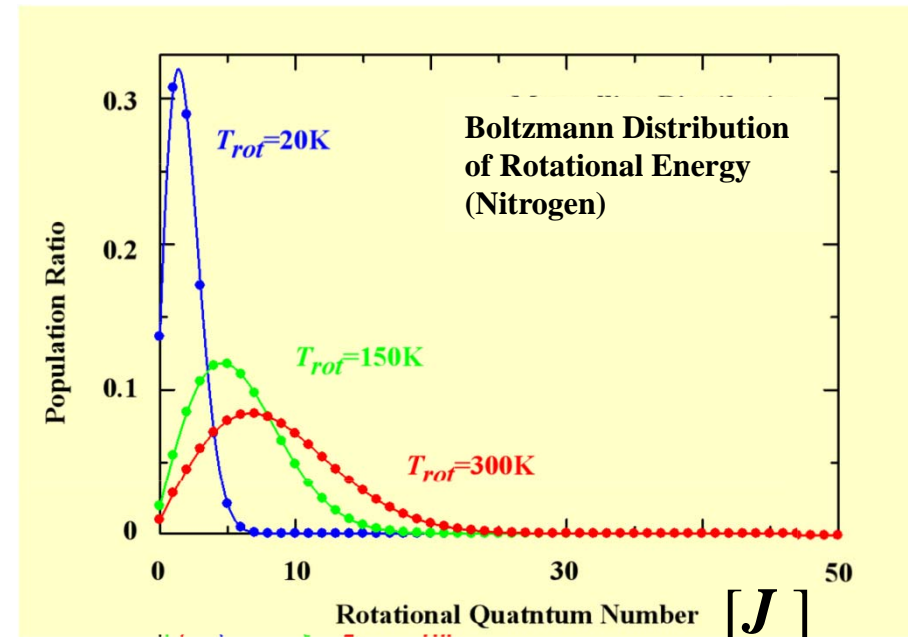
$$f(C_x) = \left(\frac{m}{2\pi k T_{tr}} \right)^{\frac{1}{2}} \exp\left(-\frac{m C_x^2}{2k T_{tr}} \right)$$

Rotational energy

$$N(J) \propto (2J + 1) \exp\left(-\frac{E_{rot}}{k T_{rot}} \right)$$



T_{tr} : Translational temp.



T_{rot} : Rotational temp.

Temperature is a determining factor of these distributions



2R+2 N₂-REMPI

Resonantly Enhanced Multi-Photon Ionization

Non-intrusive Measurement Technique with High Sensitivity and Short Response, allowing measurement of non-equilibrium phenomena in the highly rarefied gas flows

Rotational Line Intensity

$$I_{J',J''} = Cg(J'')S(J',J'')\exp(-E_{rot}/kT_{rot})$$

C: constant

$g(J'')$: nuclear spin degeneracy

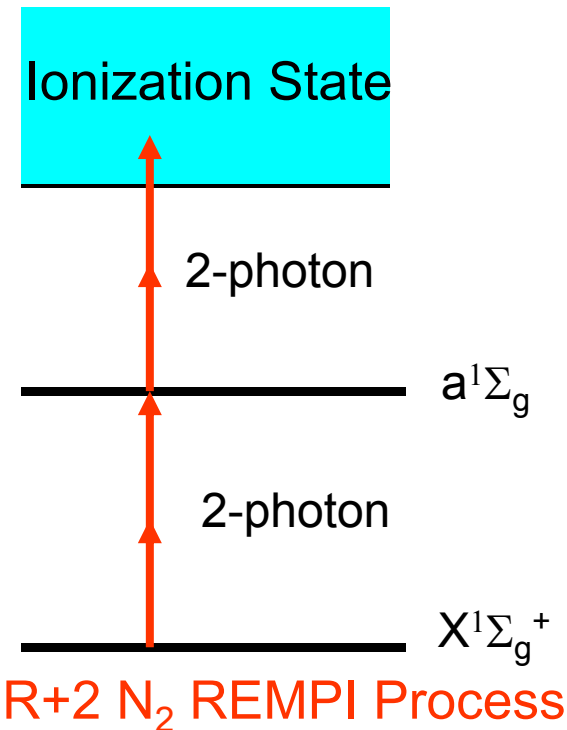
$S(J',J'')$: 2-photon

Honl-London factor

E_{rot} : rotational energy

k : Boltzmann's constant

T_{rot} : rotational temperature



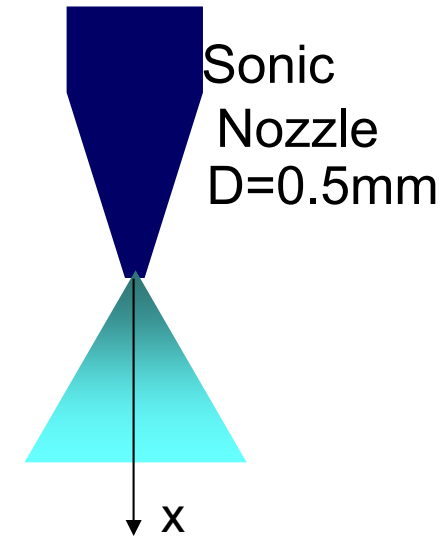
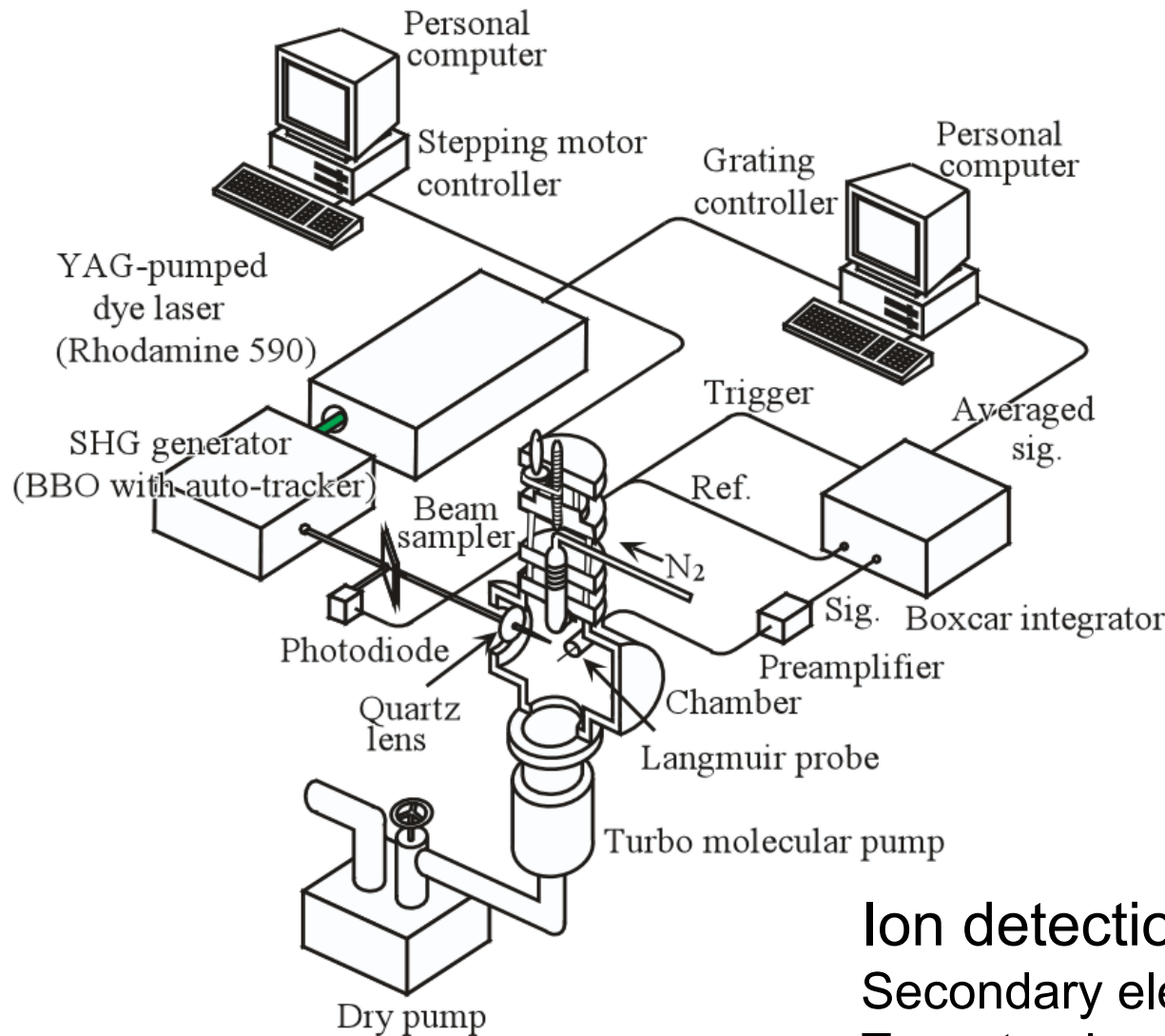
Photon energy is reduced to one-fourth of EBF

The ejected electrons by photons excite no other molecules or ions again

No consideration of the secondary electrons



Experimental Apparatus



$$P_0 = 30, 20, 10, 1 \text{ Torr}$$

$$P_b = 1.0 \times 10^{-3} - 3.3 \times 10^{-5} \text{ Torr}$$

$$T_0 = 293 \text{ K}$$

$$P_0 D = 15, 10, 5, 0.5 \text{ Torr-mm}$$

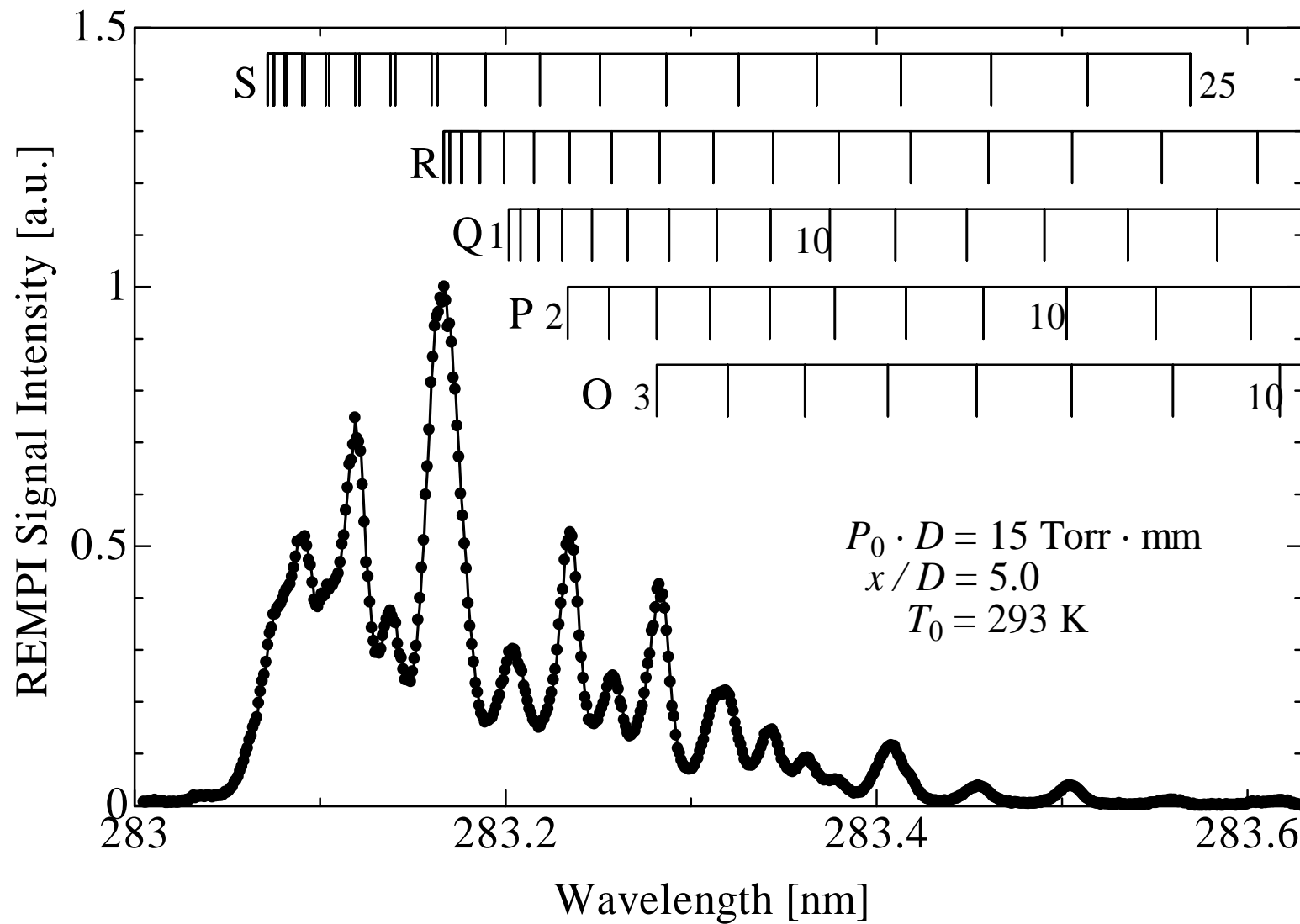
Ion detection

Secondary electron multiplier ($P_0 = 1 \text{ Torr}$)

Tungsten Langmuir probe ($P_0 \geq 10 \text{ Torr}$)



REMPI Spectrum



Boltzmann Plots

Rotational line intensity $I_{J',J''}$ in $2R+2 N_2$ -REMPI spectra

$$I_{J',J''} = Ag(J'')S(J',J'')N(J'')/(2J''+1)$$

J : rotational quantum number

(J' : resonant state J'' : ground state)

A : proportional constant independent of the rotational quantum

$N(J'')$: population number

$g(J'')$: nuclear spin statistical weight (3 and 6 for odd and even J'')

$S(J',J'')$: two-photon Hönl-London factor

$N(J'')$ is proportional to $(2J''+1)\exp(-E_{rot}/kT_{rot})$, provided that the rotational energy distribution follows the Boltzmann distribution.

$$I_{J',J''} = Ag(J'')S(J',J'')\exp(-E_{rot}/kT_{rot})$$

Rotational temperature : slope of Boltzmann plot [$\ln(I_{J',J''}/gS)$ versus E_{rot}/k],
provided to be in equilibrium.

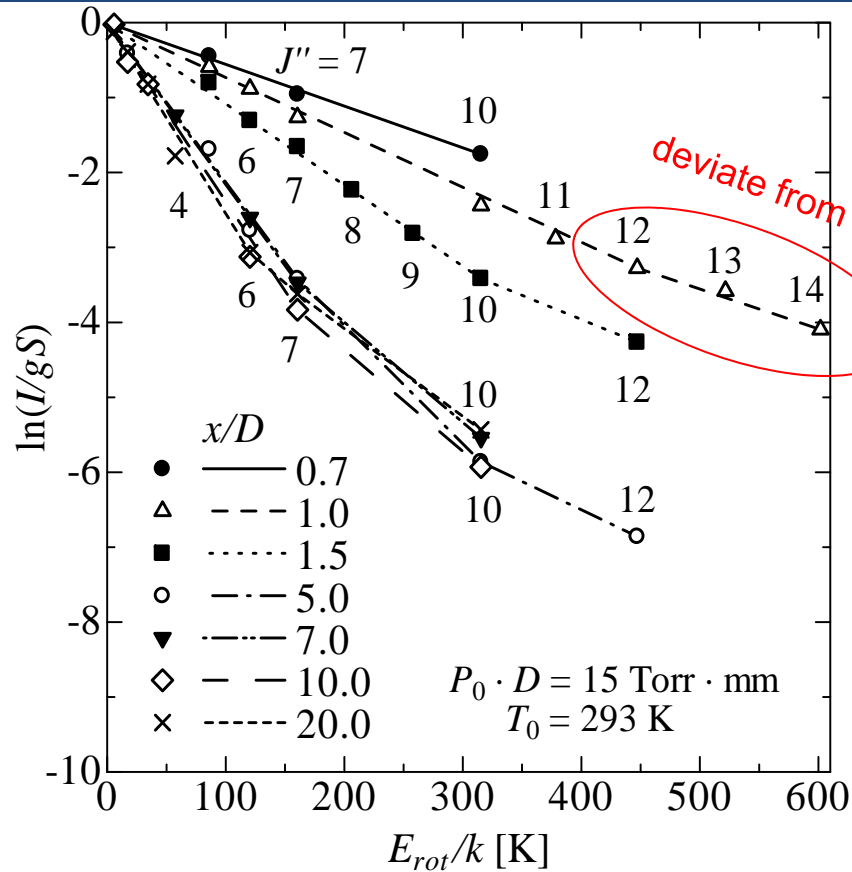
Nonlinearity in the Boltzmann plot :

Rotational energy distribution deviates from the Boltzmann distribution

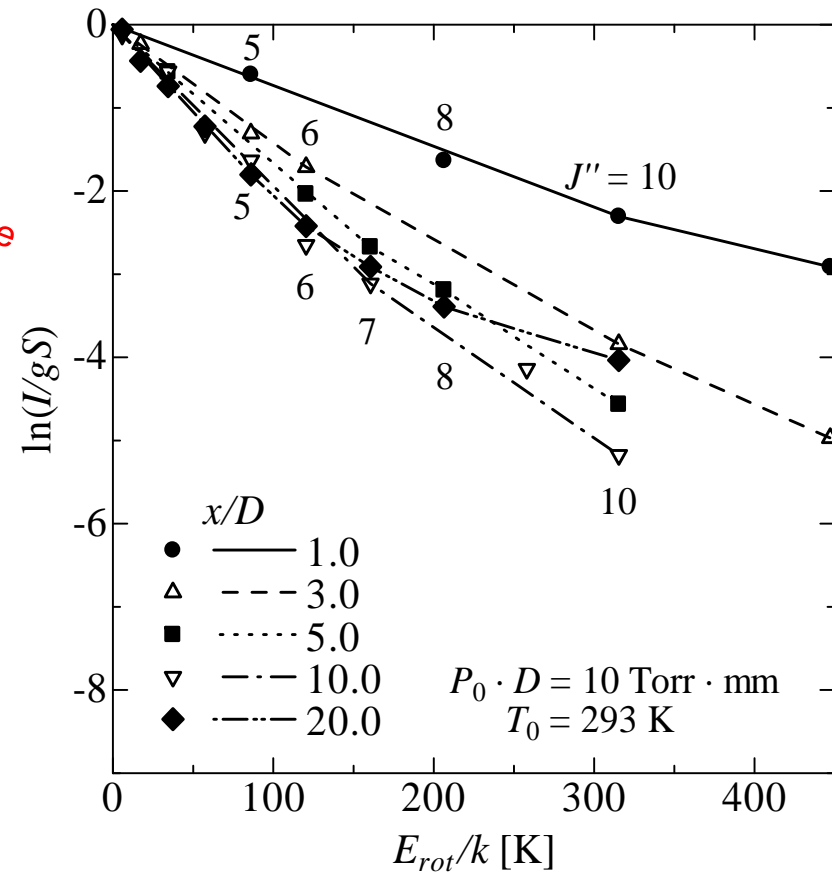
(Non-Boltzmann distribution) \rightarrow Rotational temperature cannot be defined.



Boltzmann Plots



$P_0 D = 15 \text{ Torr-mm}$

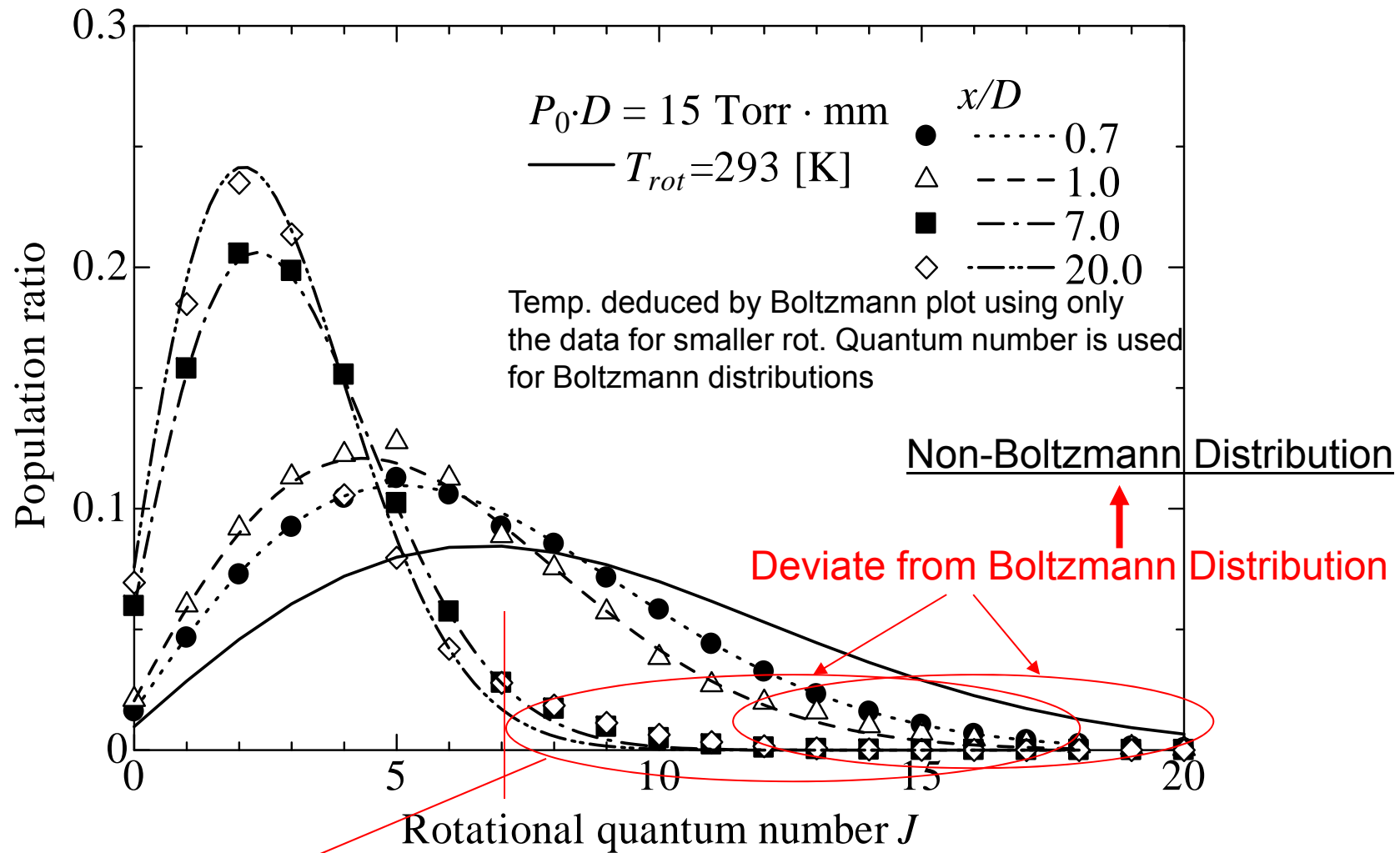


$P_0 D = 10 \text{ Torr-mm}$

Determination of the rotational temperature by using only the linear portion of the plots lying at smaller rotational quantum numbers



Rotational Energy Distribution ($P_0 \cdot D = 15 \text{ Torr} \cdot \text{mm}$)



distributions of $x/D=7.0$ and 20.0 for $J'' > 7$ shows the same tendency

→ partial freezing of the rot. population

H.Mori, T.Niimi et al., Phys. Fluids 17, 117103 (2005)



Development of PSP for Low Density Gas

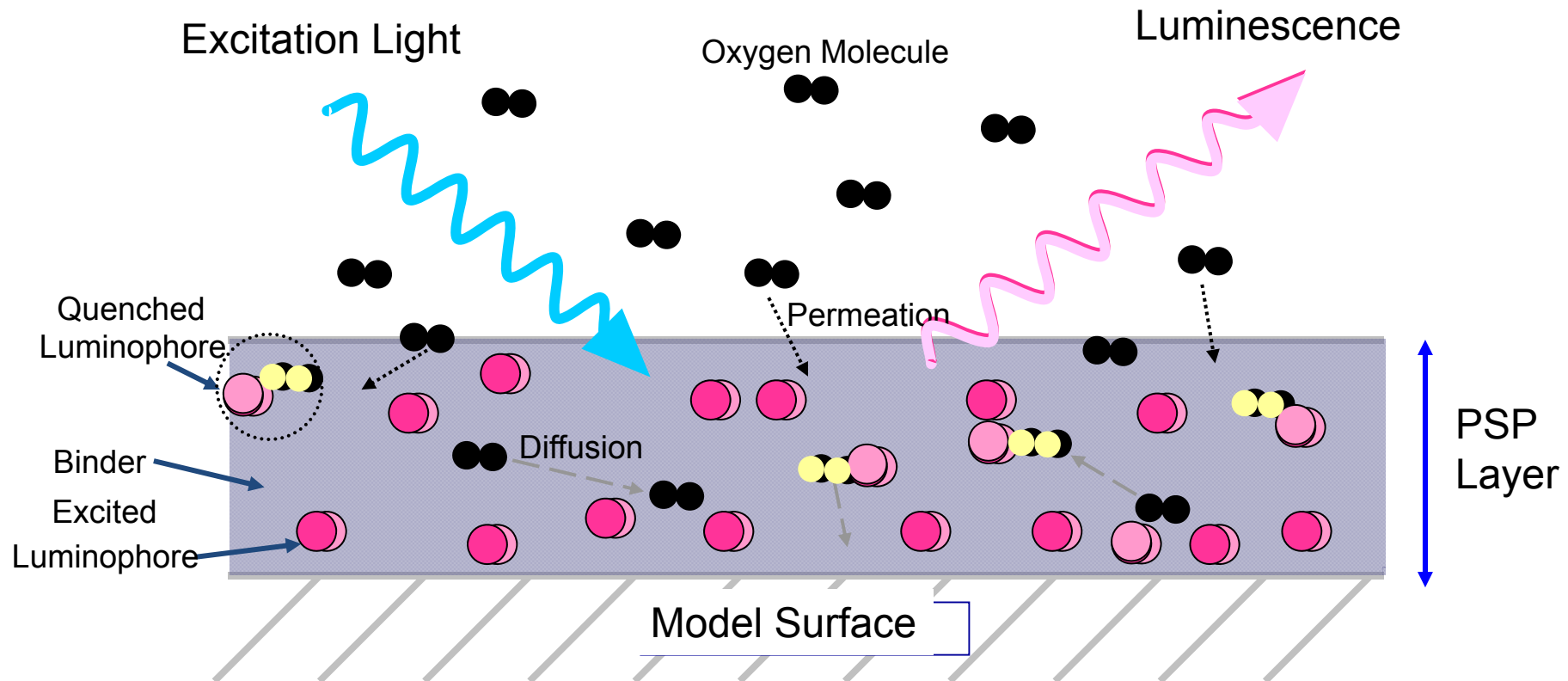


Basic 11 High Knudsen Number Flows
COE for Education and Research of Micro-Nano Mechatronics, Nagoya University

Prof. T. Niimi



Principle of PSP (Pressure Sensitive Paint)



Brighter region: Low pressure region
Darker region : High pressure region

Application of PSP in low-pressure regime is very few, because luminescence intensity does not change significantly in low- pressure range.

➔ overcome the limitation of sensitivity!



PSP in Low Pressure Regime

- Sensitivity of PSP is restricted by gas permeability inside the binder layer

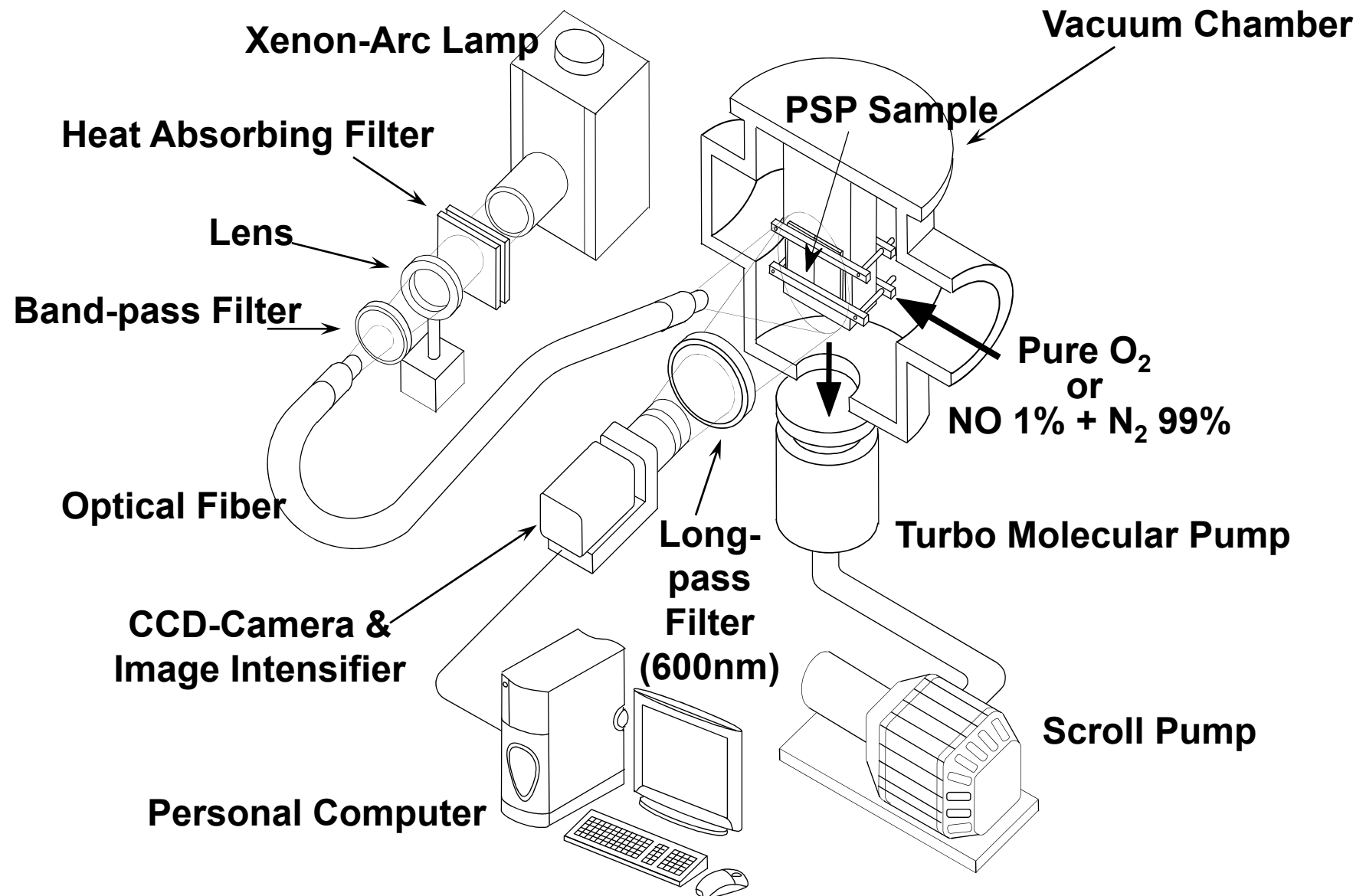
PSPs have been regarded to be inappropriate for use in low pressure regime, because change of luminescence intensity is small.

We have to overcome the limitation of sensitivity!

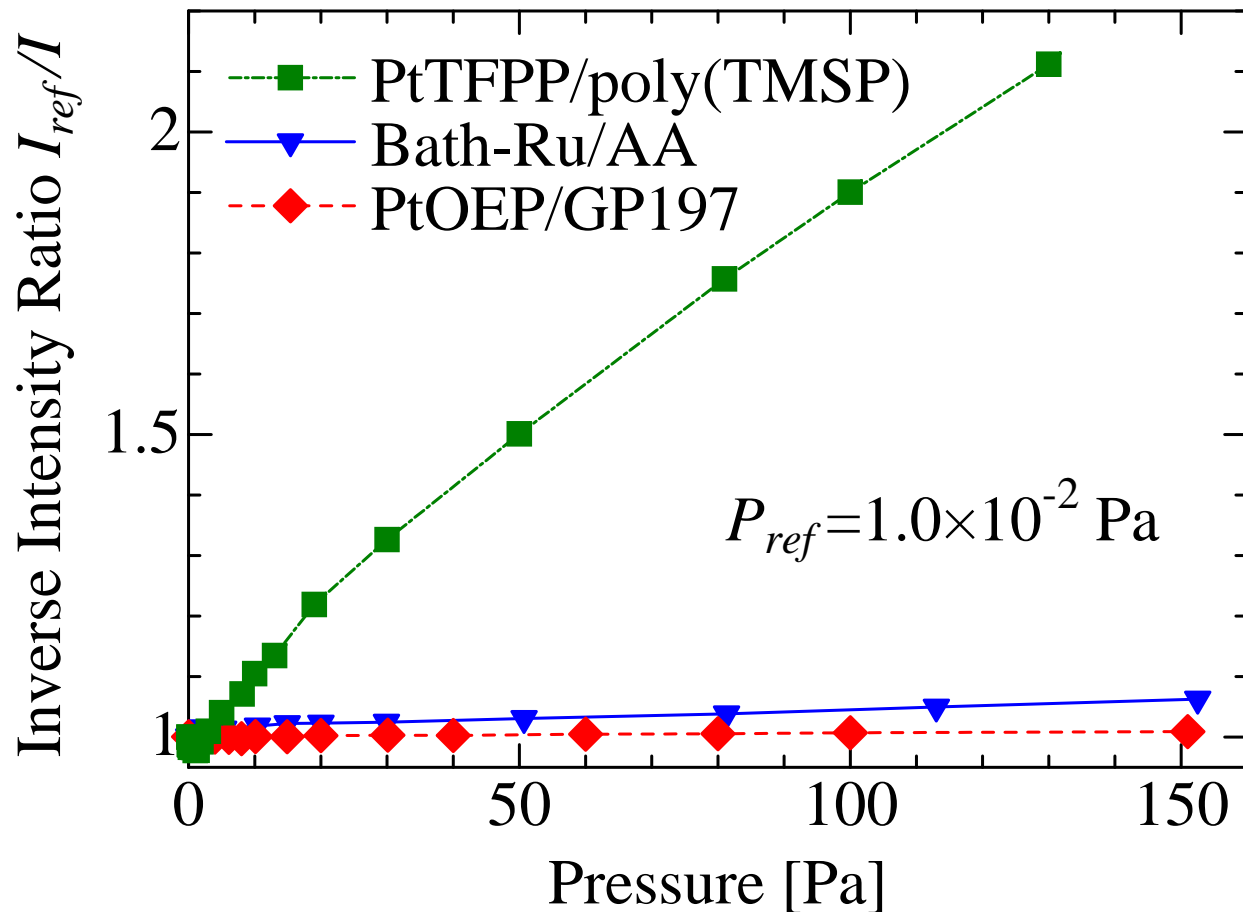
- **1st step: suitable binder?**
 - Porous surface of **anodized aluminum (AA)**: luminescent molecules are bound directly on the surface
→ **exposed** to the atmosphere : Bath-Ru/AA
 - **poly(TMSP)**: glassy polymer with **extremely high oxygen permeability** :PtTFPP/poly(TMSP)
 - (PtOEP/GP197: tested for comparison)



Experimental Apparatus



PSP in Low Pressure Regime



PSP in Low Pressure Regime

- PtTFPP/poly(TMSP): **high sensitivity**
 - ← high oxygen permeability of poly(TMSP)
- Bath-Ru/AA:
 - ▶ lower sensitivity than PtTFPP/poly(TMSP)
 - ▶ time delay for abrupt pressure change
 - ← Adsorption and desorption of oxygen molecules inside pores
- PtOEP/GP197: no sensitivity



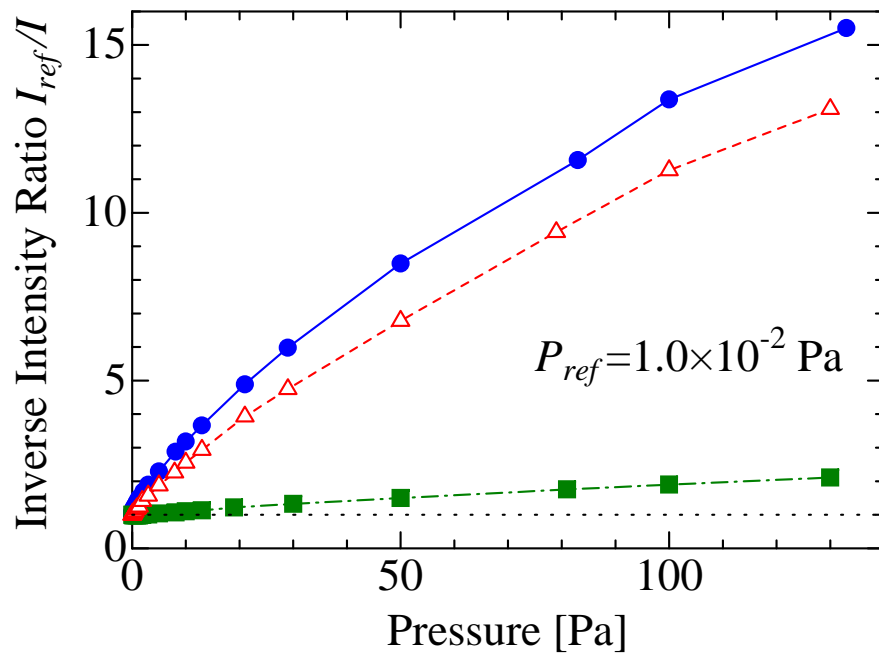
Oxygen Pressure Sensitivity of PSPs using poly(TMSP) as a Binder

- 2nd step:
Luminophore with high sensitivity to combine with the binder poly(TMSP)?
 - PtTFPP (has been used)
 - PdOEP
 - PdTFPP

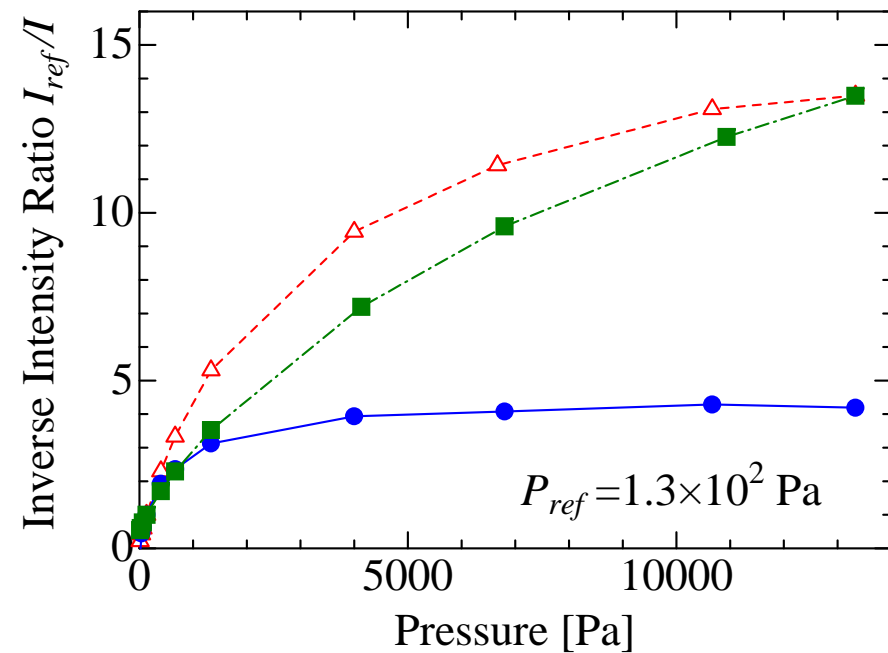


Oxygen Pressure Sensitivity of PSPs using poly(TMSP) as a Binder

✦ Luminophore: PdTFPP $\text{---}\triangle\text{---}$ PdOEP $\text{---}\bullet\text{---}$
PtTFPP $\text{---}\blacksquare\text{---}$



(a) $1.0 \times 10^{-2} - 1.3 \times 10^2$ [Pa] below 1 Torr (133.3 Pa)



(b) $1.3 \times 10^2 - 1.3 \times 10^4$ [Pa] wide range of pressure (1 – 100 Torr)



Oxygen Pressure Sensitivity of PSPs using poly(TMSP) as a Binder

- **PdOEP or PdTFPP / poly(TMSP)**
 - very powerful measurement tool at low pressure (< 1 Torr)
 - not suitable above 1 Torr (133Pa)
- **PtTFPP / poly(TMSP)**
 - useful at relatively wide pressure range (up to 100 Torr)

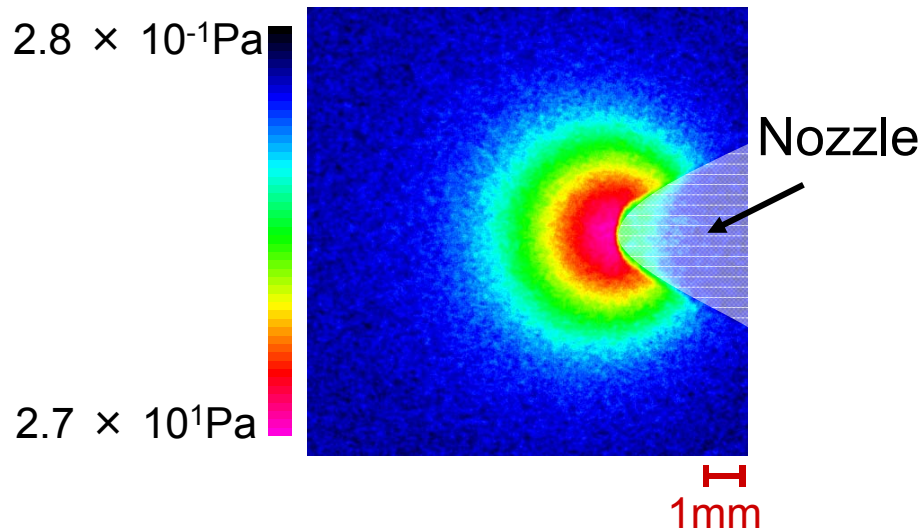
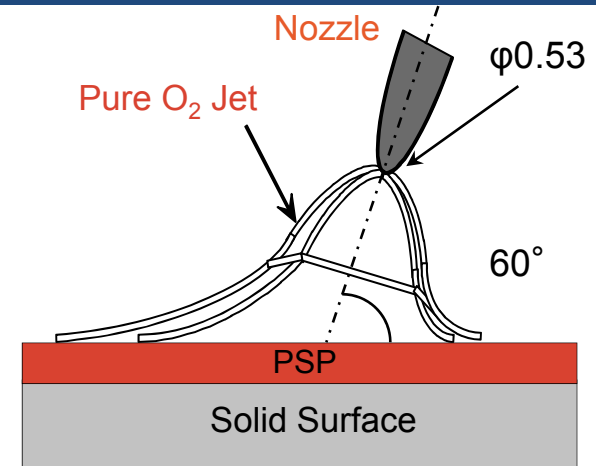


Pressure Distribution on a Solid Surface

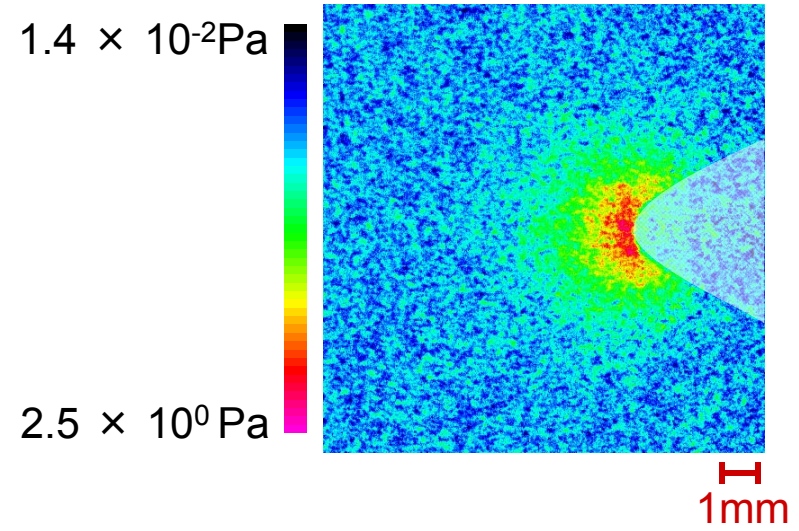
interacting with Low Density Gas Flow

★ PSP: PdOEP/poly(TMSP),
test gas: oxygen

P_s : Source Pressure
 P_b : Background Pressure



(a) $P_s = 1.3 \times 10^3 \text{ Pa}$ (10 Torr)
 $P_b = 1.3 \text{ Pa}$ (0.01 Torr)



(b) $P_s = 1.3 \times 10^2 \text{ Pa}$ (1 Torr)
 $P_b = 7.7 \times 10^{-2} \text{ Pa}$

Pressure distribution below 1 Torr is detected !



Development of PSMF for Micro- and Nano-Devices (PSMF: Pressure Sensitive Molecular Film)

LB Method for Fabrication of PSMF Application of PSMF to Micro-Flow



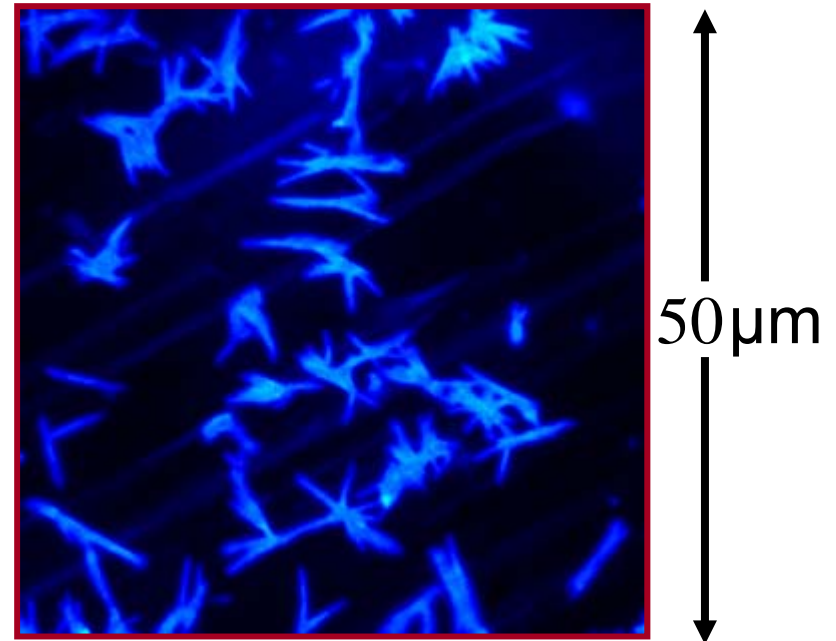
Conventional PSP

Some problems for

High Knudsen Number Flows

(the flow fields around micro-
or nano-devices)

- Large thickness ($>5\mu\text{m}$)
- Large surface roughness ($\sim\mu\text{m}$)
- Low spatial resolution due to **aggregation** of luminescent molecules



Aggregation of luminescent molecules in the layer

For High Knudsen Number Flows:

PSMF (Pressure Sensitive Molecular Film)

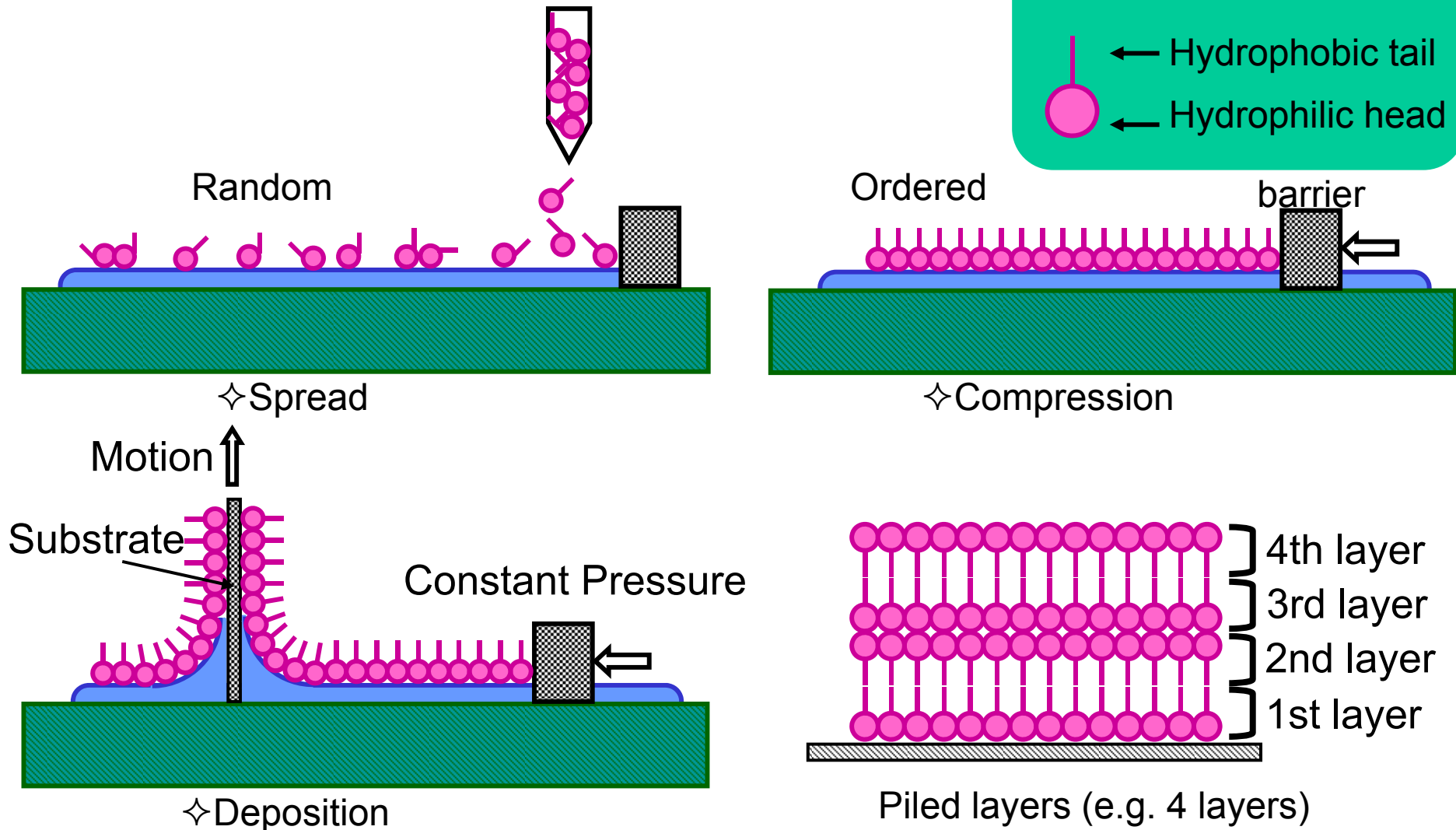
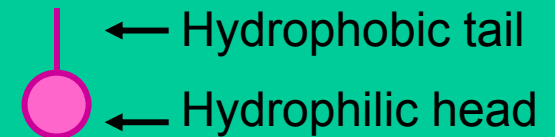
with nanometer order thickness and ordered molecular structure



Langmuir-Blodgett Method

To fabricate the thin film with nanometer order

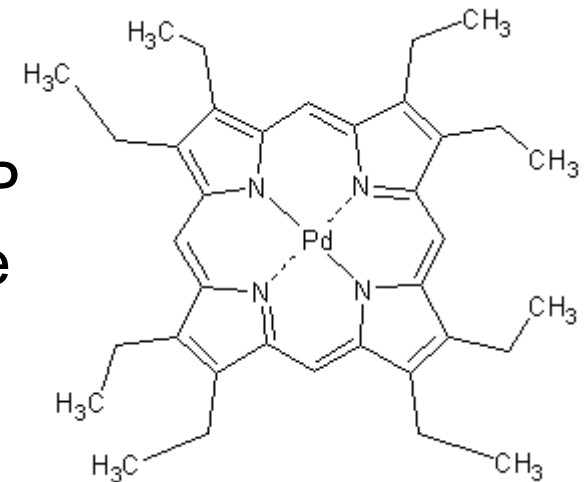
Amphiphilic molecule



Component of PSMF

- **PdOEP (Pd(II) Octaethylporphine)**

- conventional PSP composed of PdOEP
- high sensitivity in low pressure regime
- hydrophobic molecule
- difficult to fabricate a stable LB film

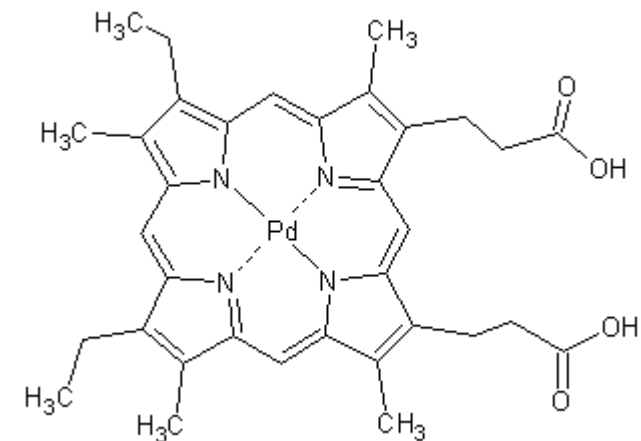


PdOEP

- **PdMP (Pd(II) Mesoporphyrin IX)**

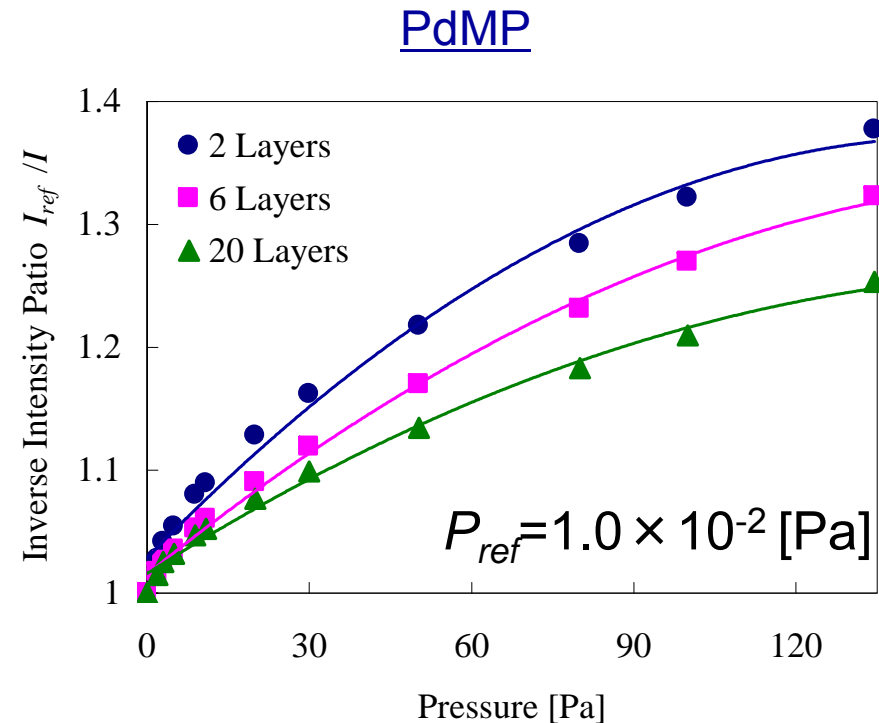
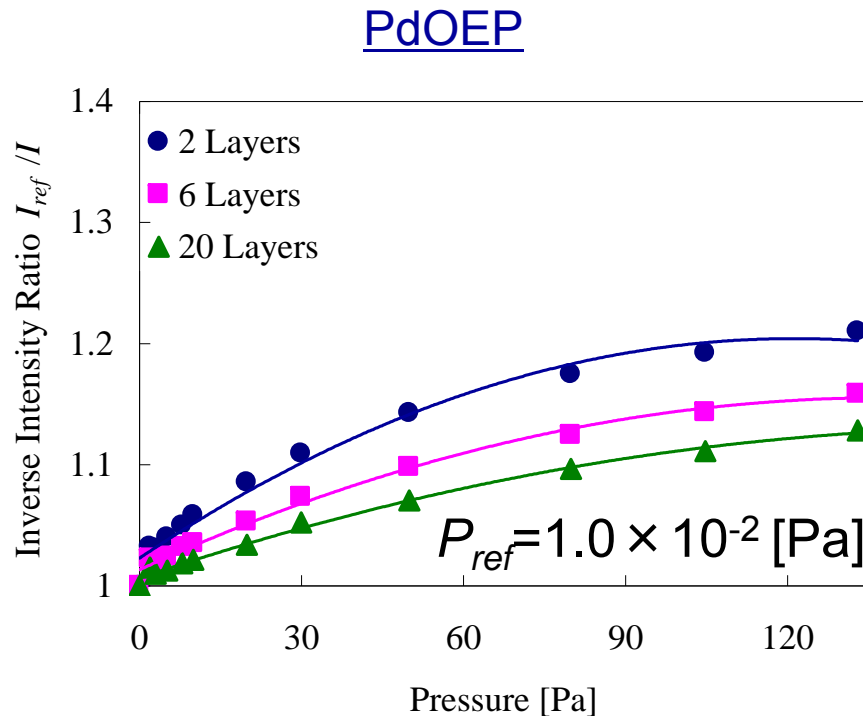
- amphiphilic molecule
- stable LB film can be obtained

prepare three types of samples
2,6 and 20 layers of PSMF to
test their pressure sensitivity



PdMP

Pressure Sensitivity



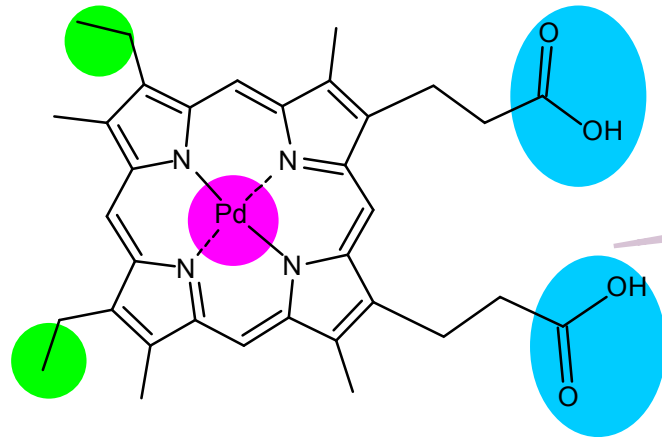
PSMF composed of PdMP has higher sensitivity than that of PdOEP
The sensitivity of 2-layer PSMF is higher than the others

PSMF has sufficient sensitivity in the low pressure regime
with high Knudsen number

Y. Matsuda et al., *Experiments in Fluids* Vol.42, No.4, pp.543-550 (2007)



PSMF with Higher Pressure Sensitivity in Atmospheric Condition



Pd(II)Mesoporphyrin IX
(PdMP)

Applicable Pressure Range
<130Pa (pure O₂)

MEMS Device
used in Atmospheric Condition

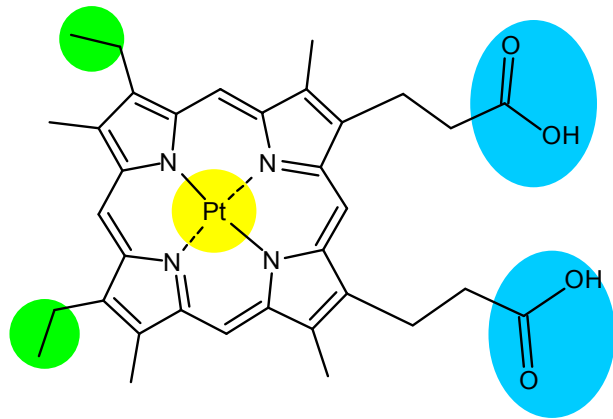


Develop New PSMF
with Higher Pressure Sensitivity
in Atmospheric Condition

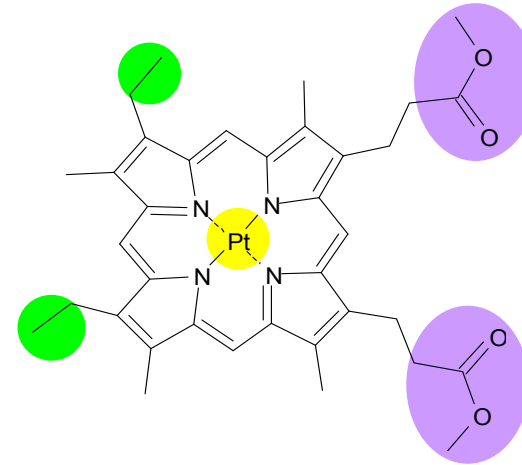
Candidates of luminescent molecule for PSMF
PtMP, PtMP-DME, PtPP, CuMP-DME



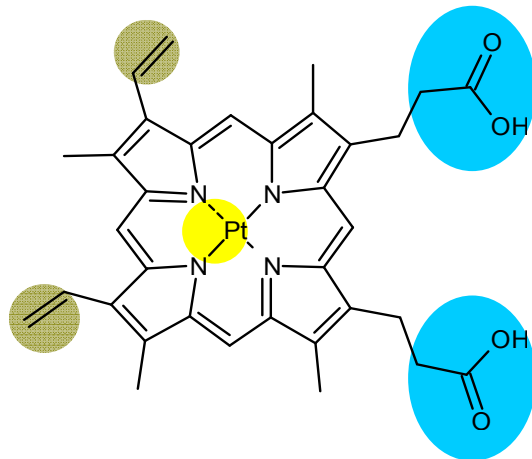
Luminescent Molecules



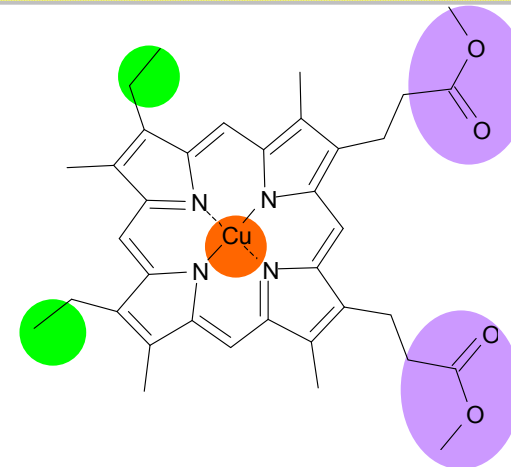
Pt(II)Mesoporphyrin IX (PtMP)



Pt(II)Mesoporphyrin IX-DME (PtMP-DME)



Pt(II)Protoporphyrin IX (PtPP)

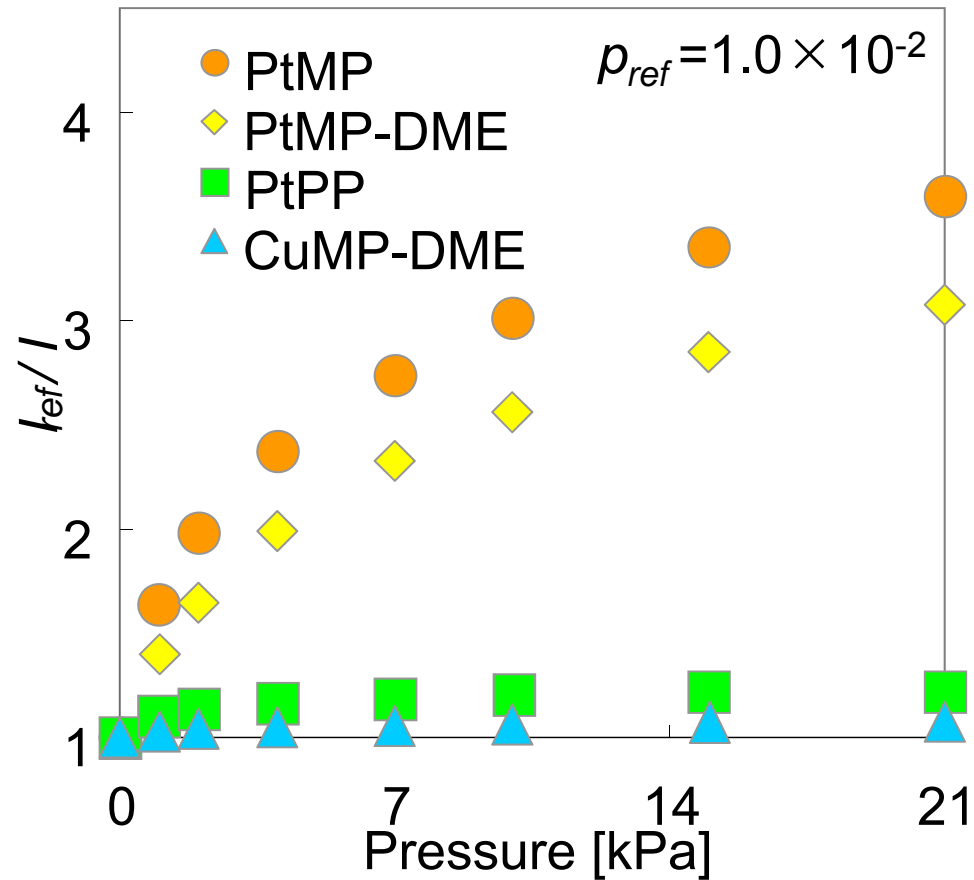


Cu(II)Mesoporphyrin IX-DME (CuMP-DME)

→ prepared 6-layers of PSMF for each luminescent molecule to test the pressure sensitivity in the range of pressure up to 21 kPa, partial pressure of oxygen in air



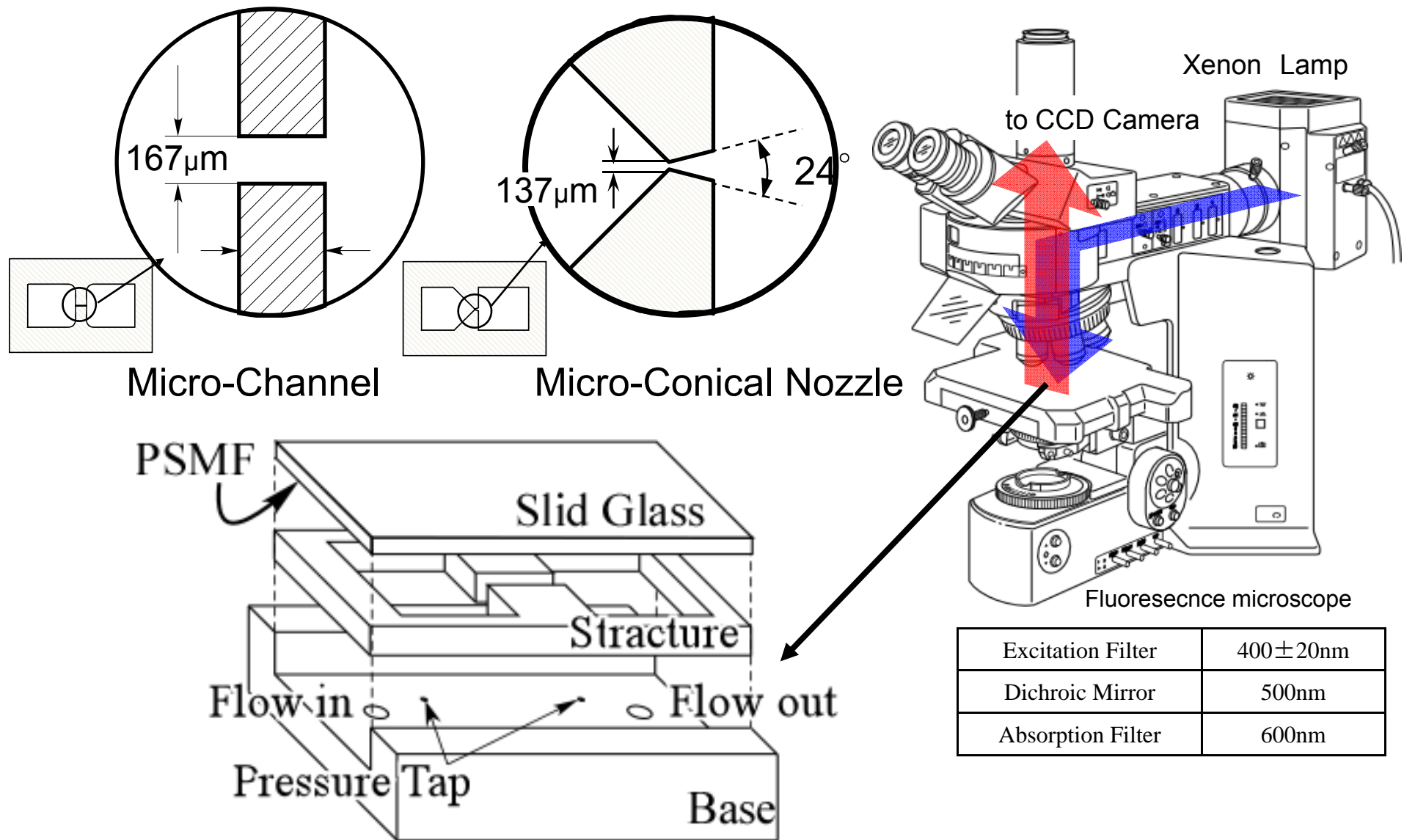
Stern-Volmer plot



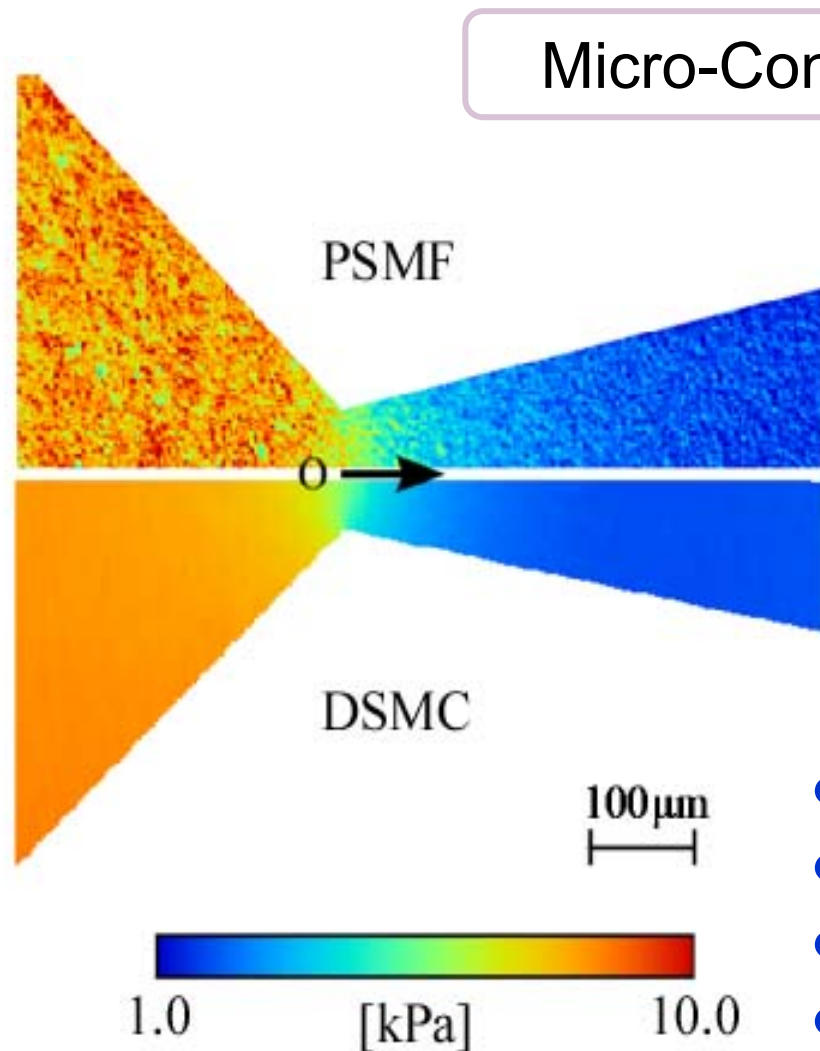
PtMP has the highest pressure sensitivity below 21kPa



Applications of PSMF to Micro-Scale Channel and Nozzle



Pressure distribution measured by PSMF



- PSMF ••• PSMF-PtMP 6-layer
- Gas ••• O₂ 21.0%, N₂ balance
- Source pressure ••• 10.0kPa
- Back pressure ••• 1.0kPa

Y. Matsuda et al, Microfluidics and Nanofluidics, 10, No.1, pp.165-171, 2011



Rotating Disks

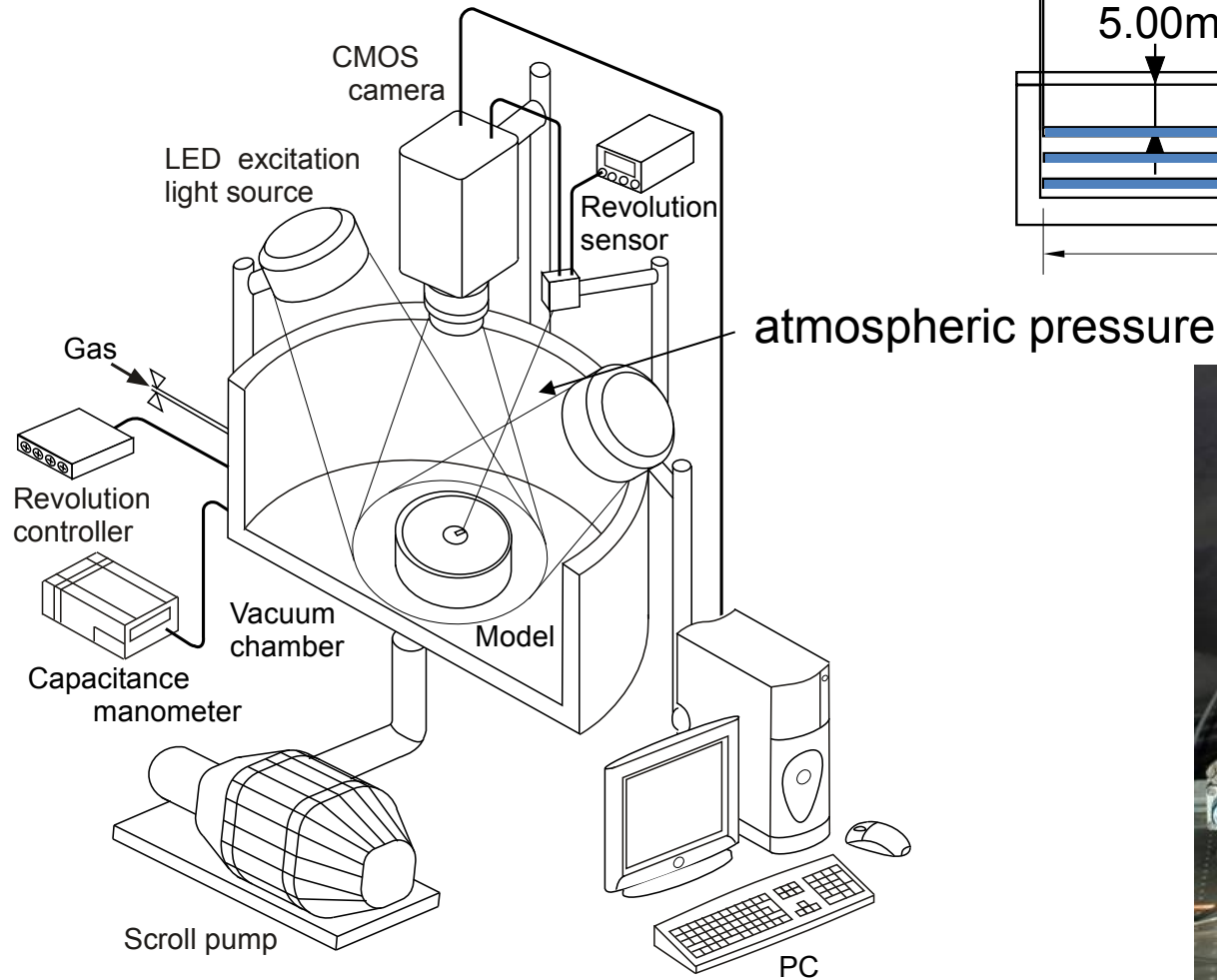
Surface Pressure Distribution on Rotating Disks

Mixing Chamber

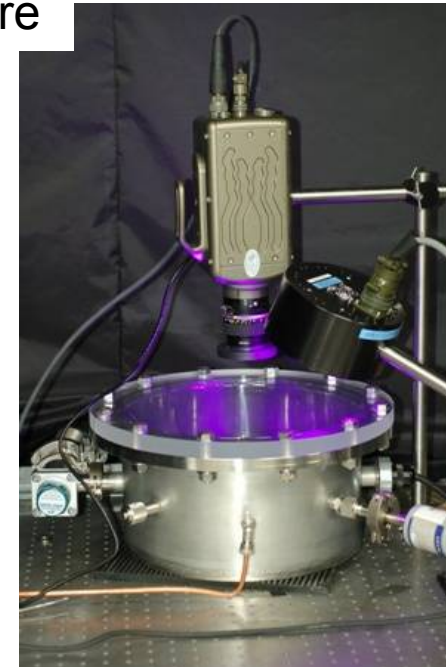
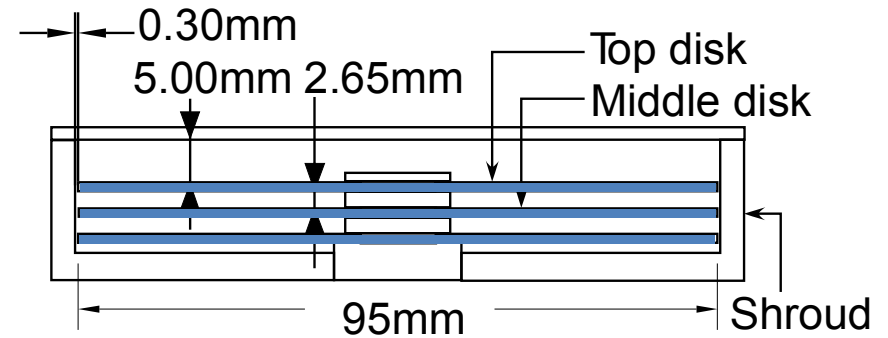
Density Fluctuation at Interface of Interacting Parallel Two Jets



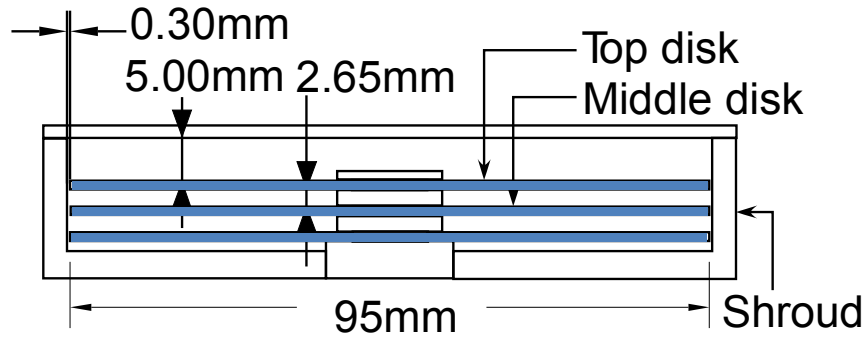
Experimental Setup



Three Disks
PSP is painted on Top and Middle Disks

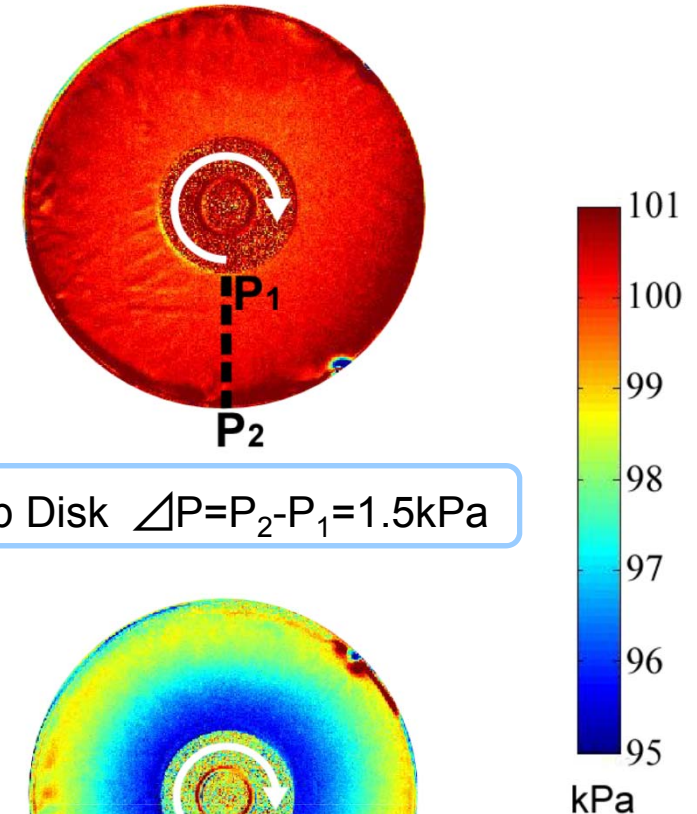
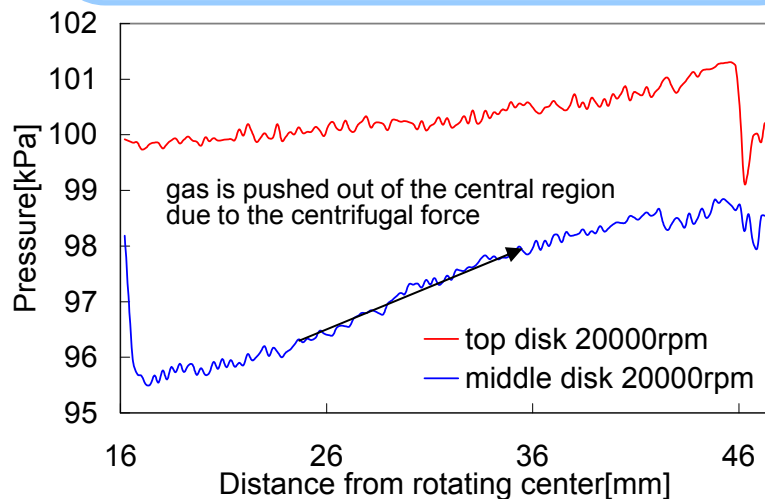


Surface Pressure Distribution on Rotating Disks



Experimental conditions

Disk Diameter : 95mm
 PSP: PtTFPP/poly(TMSP), TiO_2 2.0g/l
 Carrier Gas : N_2 79%, O_2 21%
 Rotating Speed (I) : 20000rpm
 Rotating Speed of Reference Image (I_{ref}) : 300rpm
 Integration Time:20ms
 Ambient Pressure : 100kPa

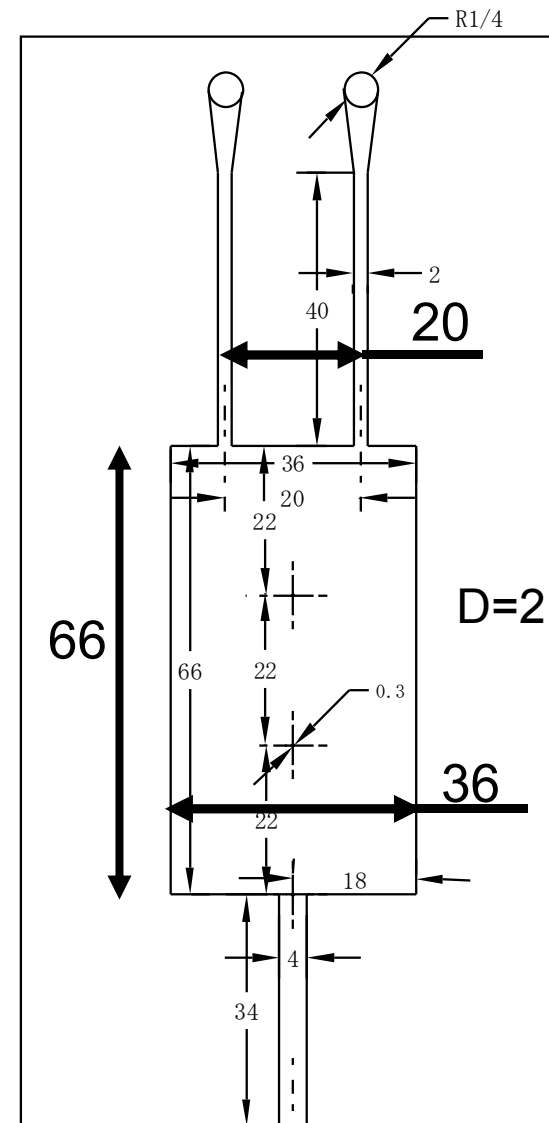
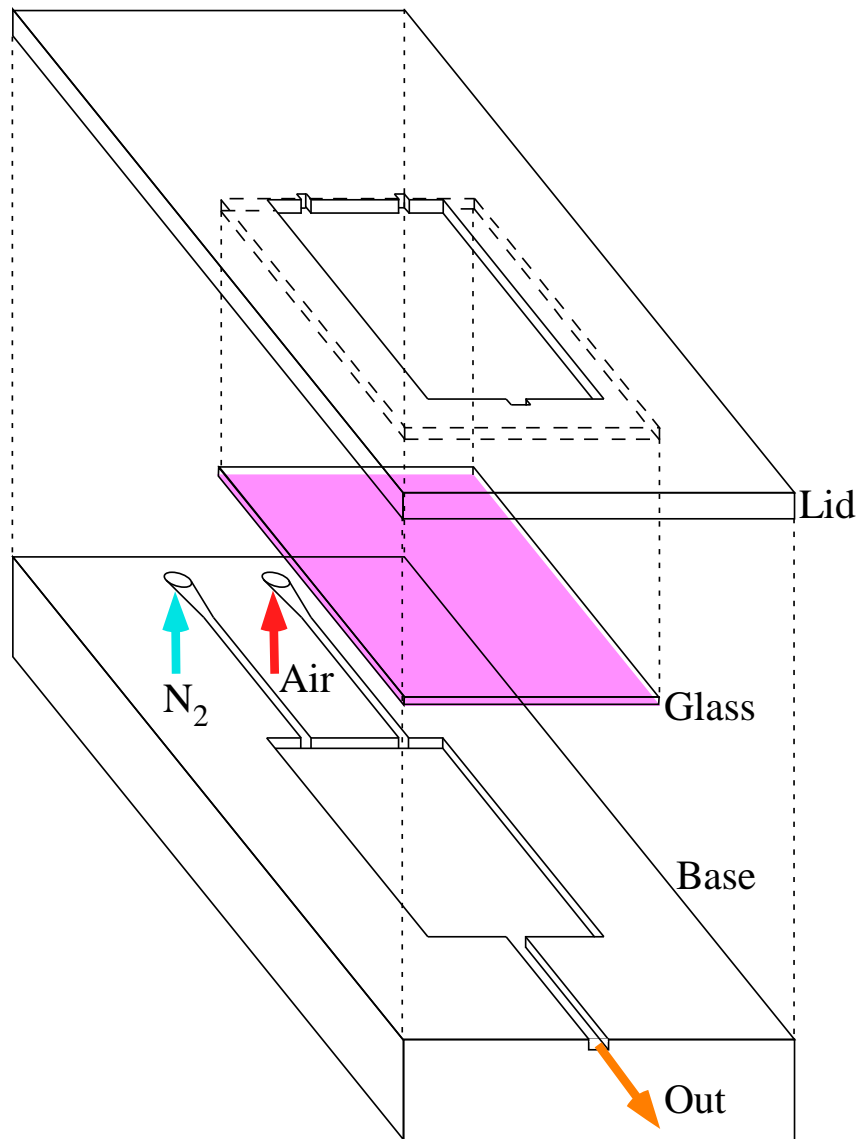


Mixing Chamber

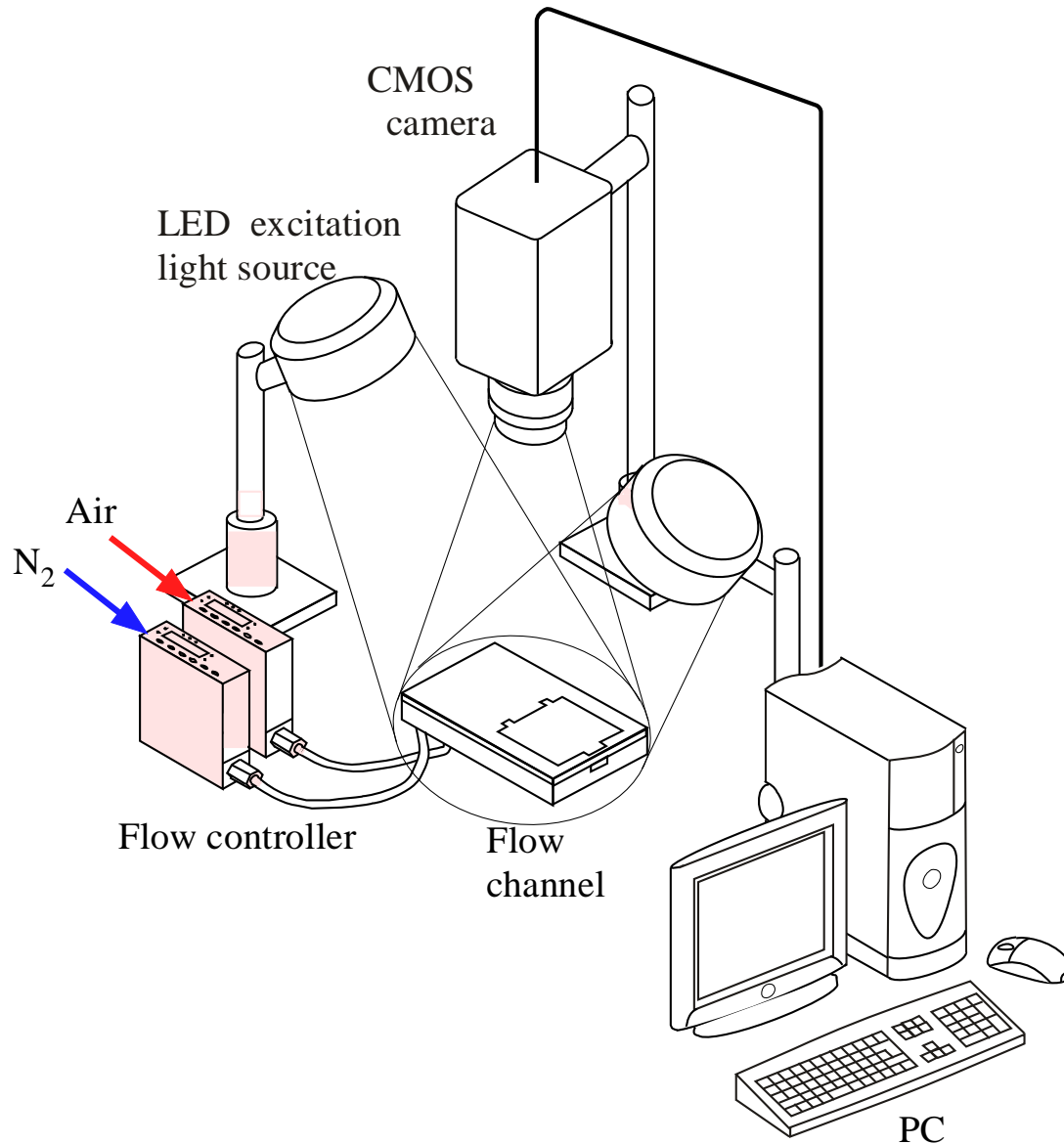
Density Fluctuation at Interface of Interacting Parallel Two Jets



Mixing Chamber



Experimental Setup



PSP

PtTFPP/poly(TMSP)

- Relatively high sensitivity around atmospheric pressure
- High Oxygen Permeability,
- Quick Time Response

Flow controller

Digital Mass Flow Controller MQV0002(Yamatake Co. Ltd.)

Flow Rate : 20ml/min~2000ml/min

Accuracy : ± 10 ml/min

(Q=20~1000ml/min)

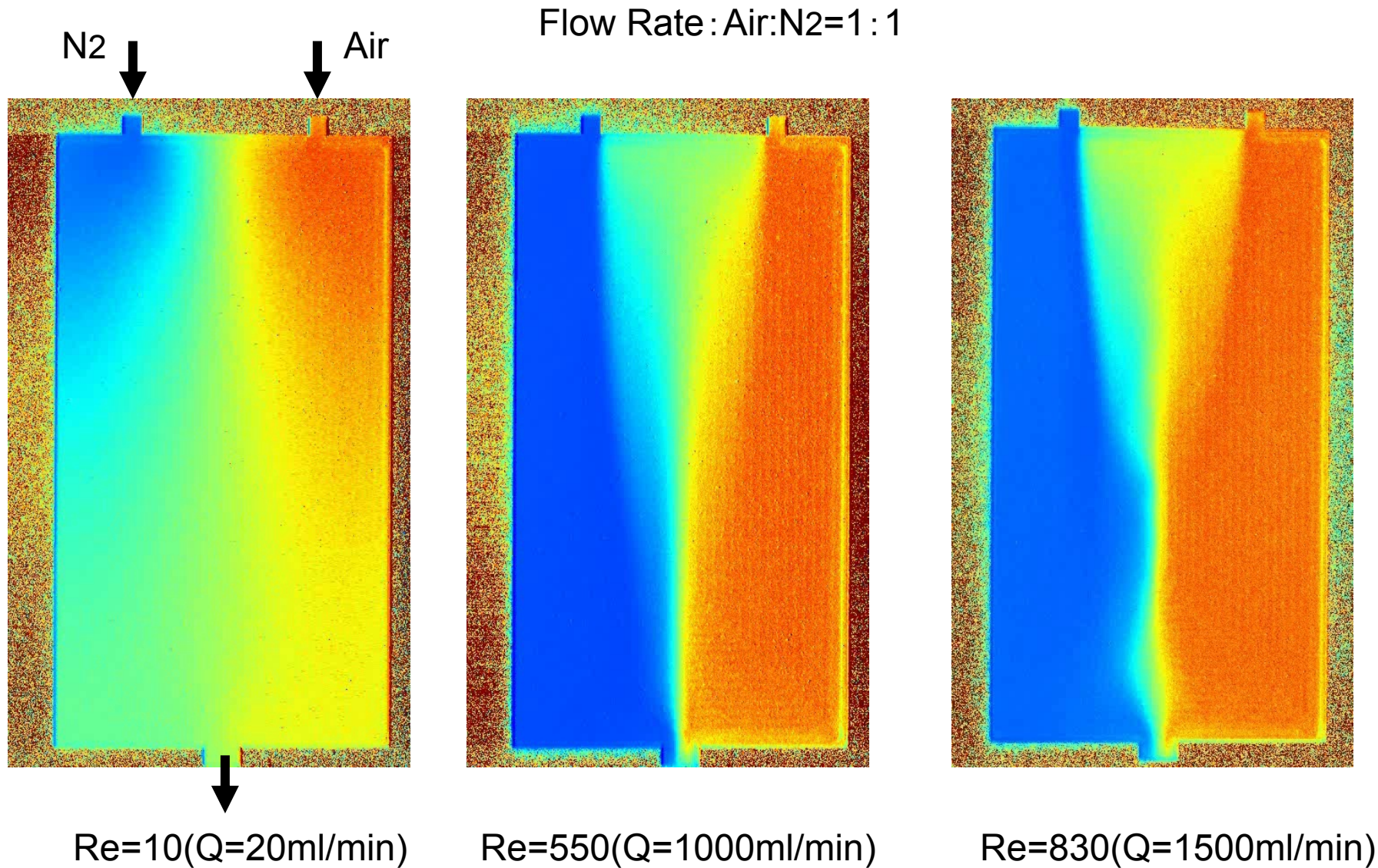
± 20 ml/min

(Q=1000~2000ml/min)

Response Time : 0.3s



Experimental Results



Summary

1. High Knudsen Number Flows
2. Development of Optical Diagnostics Methods for High Knudsen Number Flows
 - Laser Induced Fluorescence (LIF) :
 - Resonantly Enhanced Multi-Photon Ionization (REMPI)
 - Rotational Nonequilibrium in Low Density N₂ Jets
3. Application of Pressure Sensitive Paint (PSP)
 - to High Knudsen Number Flows
 - Applications of PSP to Low Density Gas Flows,
 - Development of PSMF for Micro- and Nano-Devices
(PSMF: Pressure Sensitive Molecular Film)
 - LB Method for Fabrication of PSMF
 - Application of PSMF to Micro-Flows
4. Applications of PSP in Atmospheric Condition
 - Rotating Disks
 - Mixing Chamber

